

HEAT EXCHANGER SPECIFICATION SHEET

Customer	KP Engineering	P.O. No.	J144-B-007
Address	Tyler, TX	Job No.	1203-04
Plant Location	Holly Frontier- Tulsa, OK	Date	2/24/16
Service of Unit	Debutanizer Steam Reboiler	Item No.	374-E-C221
Size	40-240	Type	AEU Horizontal Connected In 1 Parallel 1 Series
Surf/Unit (Eff), Sq Ft	3189	Shell/Unit	1 Surf/Shell (Eff) 3189 Sq Ft

PERFORMANCE OF ONE UNIT

Fluid Allocation		Shell Side		Tube Side			
Fluid Name		Steam		Debutanizer Bottoms			
Fluid Quantity, Total lb/hr		32,535		615,692			
Vapor (In/Out)							206,934
Liquid				615,692			408,758
Steam		32,535					
Water			32,535				
Noncondensable							
Temperature (In/Out) F		552.70	420.79	334.40			354.6
Density			52.71	32.27			31.67
Viscosity cP		0.0196	0.1239	0.1000	0.011	V/L	0.095
Molecular Weight, Vapor		18.02					83.87
Molecular Weight, Noncondensable							
Specific Heat Btu/lb-F		0.511	1.200	0.733	0.598	V/L	0.758
Thermal Conductivity Btu/hr-ft-F		0.026	0.377	0.038	0.017	V/L	0.036
Latent Heat Btu/lb							
Inlet Pressure psia			314.70				206.70
Velocity ft/sec							11.21
Pressure Drop, Allow/Calc psi		10.000	0.765	15.000			5.551
Fouling Resistance (min) ft-hr-F/Btu			0.00050				0.0040
Heat Exchanged Btu/hr		28,750,000	MTD (Corrected)	72.7	F		
Transfer Rate, Service		124.0 Btu/ft ² -hr-F					

CONSTRUCTION OF ONE SHELL

		Shell Side		Tube Side	
Design Pressure PSI		675 / F.V.		525 / F.V.	
Test Pressure		CODE		CODE	
Design Temperature / MDMT F		725 / 300 / -20		425 / 300 / -20	
No Passes per Shell		1		2	
Corrosion Allowance inch		0.1250		0.1250	
Connections	In inch	1 @ 12	600 # RFWN	1 @ 10	300 # RFWN
	Out inch	1 @ 6	600 # RFWN	1 @ 14	300 # RFWN
Tube No	313U OD	1.0000	Thk(Min)	0.109	inch
Tube Type	BARE			Length	20.0 feet
				Pitch	1.250 inch
				Layout	45
				Material	SA-214
Shell	SA-516-70	ID	40.00	inch	Shell Cover SA-516-70
Channel or Bonnet	SA-516-70				Channel Cover SA-516-70
Tubesheet-Stationary	SA-350-LF2				Tubesheet-Floating
Floating Head Cover					Impingement Plate Circular plate
Baffles-Cross	SA-36	Type	Sing Seg - Vert	%Cut (Diam)	30.10
				Spacing(c/c)	9
Baffles-Long					
Supports-Tube	SA-36		U-Bend	SA-36	Type Full
Bypass Seal Arrangement			Tube-Tubesheet Joint	Rolled & Double Grooved	
Gaskets - Shell	GMGC		Tube	GMGC	
-Floating Head					
Code Requirements			ASME Section VIII, Div. I	TEMA Class	R, API 660

Remarks: Unit is thermally, vibrationally and mechanically guaranteed by Taylor Forge Engineered Systems. Two grounding lugs. Counter current flow. National Board registration included. 1" minimum bolting. Two sets of spare gaskets included. Lath Stud type fireproofing supports. PMI bolting. Full X-ray.

Revised: Gasket material per customer by K Duke

Design Verification

Doc. No.:	1203-04-K01	Doc. Title:	Mechanical Design Calculations		
TFES Job No.:	1203	Customer:	KP Engineering		
Item No.(s):	04	P.O. No.:	J1422-B-007		
No. of Pages:	53 Pages plus cover	Customer Item No.(s):	374-E-C221		
Job Description:	One 40"-240" AEU Debutanizer Steam Reboiler				
CODE:	ASME - B31.1 <input type="checkbox"/>	ASME - Section I	<input type="checkbox"/>		
	ASME - B31.3 <input type="checkbox"/>	ASME - Section VIII Div. 1	<input checked="" type="checkbox"/>		
	ASME - B31.4 <input type="checkbox"/>	ASME - Section VIII Div. 2	<input type="checkbox"/>		
	ASME - B31.8 <input type="checkbox"/>	NBIC	<input type="checkbox"/>		
	ASME - B31.11 <input type="checkbox"/>	Other _____	<input type="checkbox"/>		
Customer Comments / Stamp:					
7	Matls corrected to as-built (ie. Normalized)	KAD	ADR	07/05/16	
6	Added PWHT to Channel (pg 1)	KAD	ADR	04/21/16	
5	Added stress ratio to test calc	KAD	ADR	03/14/16	
4	Rev gasket material	KAD	ADR	02/24/16	
3	Rev Support calculations, Tubesheet calcs with strength weld	KAD	ADR	02/05/16	
2	Tubesheet matl. from plate to forging. Shell flange forging matl. from SA-105 to SA-350-LF2. Added Test @ 1.3xMAWP	KAD	ADR	01/22/16	
1	Rev per customer comments	KAD	ADR	12/29/15	
0	Original Issue	KAD	ADR	12/03/15	
Rev.	Description	Prepared	Approved	Date	

P.O. no: J1422-B-007		Job no : 1203-04	
Date: 07/05/16 By: K Duke		Item no: 374-E-C221	
Customer: KP ENGINEERING		HORIZONTAL	
Service : DEBUTANIZER STEAM REBOILER		U-TUBE	
No. Shells: 1 Par. 1 Ser. 1		Surface: 3189 Each 3189 TotalSq.Ft.	
Size: 40 x 240 " Type: A E U			
313 Tubes x 1.000 OD x 0.109 Thk BWG x 240 " Lg 1.2500 ROT.SO. Pitch			
Tubes : SA-214			
Shell : SA-516-70N			
Channel : SA-516-70			
Tubesheets : SA-350-LF2CL1			
Cross Baffles : SA-36		No. 23 Pitch 9.000 Cut 0.30 SGL SEG VERT	
Code requirements: SHELL SECT VIII STAMP YES CHANNEL SECT VIII STAMP YES			
Natl. Board #YES		Canadian Reg. # Tema R	
Weight Ea Dry: 38550 Lbs		Wet 50250 Lbs Bundle 18500 Lbs	
Shell		Tubes	
Design Press/Ext		675.0 / 15.0 psig 525.0 / 15.0 psig	
Test Press.(1.3x Des x Sr)		1067.0 psig 684.0 psig	
Test Press.(1.3x MAWP x Sr)		1067.0 psig 684.0 psig	
Sr = Stress ratio		1.216 psig 1.0 psig	
Design Temp/MDMT		725 / -20 deg.F 425 / -20 deg.F	
Corrosion Allowance		0.1250" 0.1250"	
No. of Passes		1 2	
Radiograph		FULL FULL	
Stress Relieve		NONE deg.F 1150 / U-Bends 1200 deg.F	
Paint: PREP & PAINT PER CUSTOMER SPECIFICATIONS			
INSUL BY OTHERS. INSUL SUPPORT BY TFES.			
Remarks: TUBE JOINT: EXPANDED & DOUBLE GROOVED.			
PMI REQUIRED: FLG STUDS			
NOZZLES			
Label	Size & Type	Bore Sch	Proj Weld Pad
S1 Shl INLET	12.00"- 600#RF WN	11.374 80	40.000 18.500x 1.000"
S2 Shl OUTLET	6.00"- 600#RF WN	5.761 XH	40.000 9.625x 0.875"
T1 Fch INLET	10.00"- 300#RF WN	9.062 120	40.000 0.000x 0.000"
T2 Fch OUTLET	14.00"- 300#RF WN	12.124 100	40.000 0.000x 0.000"

PRESSURE VESSEL CALCULATIONS PER RCS SHELL AND TUBE DESIGN SOFTWARE Version : 01.01.16
 SUPPORT / SEISMIC & WIND CALCULATIONS PER PV ELITE SOFTWARE Version : 17.00.00.000

SHELL SIDE

```

=====
NOM.
SIZE          PRESS.      TEMP.        C.A.         RT    H.T.        TEMA
40.000"      675.0psig    725deg.F    0.1250"     FULL    0deg.F     R
=====

```

ASME CODE SEC.VIII DIV.I UG-27 [CYLINDER]

===DESIGN CONDITIONS=====CYLINDER CALCULATIONS ON I.D. FORMULA===
SHELL CYL

MATERIAL: SA-516-70N PRESS = 675.0psig TEMP = 725deg.F

I.D. NOM. THK. RW=0/PIPE=1 CORR.ALLOW EFFICIENCY

40.0000" 1.2500" 0 0.1250" 1.00

ATM.STRS DES.STRS STRS/STRS MAWP N-C MAWP CORR.

20000.0psi 16450.0psi 1.216 1204.819psig 889.724psig

P*(R+CA)

----- = 0.84664" + CORR. ALLOW. = 0.97164" + MTOL. = 0.97164"

S*E-(.6*P)

ASME CODE SEC.VIII DIV.I UG-32 [ELLIPTICAL HEAD]

===DESIGN CONDITIONS=====ELLIPTICAL HEAD CALCULATIONS ON I.D. FORMULA===
SHELL HEAD

MATERIAL: SA-516-70N PRESS = 675.0psig TEMP = 725deg.F

I.D. (min)THK. CORR.ALLOW EFFICIENCY

40.0000" 1.0000" 0.1250" 1.00

ATM.STRS DES.STRS STRS/STRS MAWP N-C MAWP CORR.

20000.0psi 16450.0psi 1.216 995.025psig 712.121psig

P*(R+CA) 0.0000

----- = 0.82920" + CORR. ALLOW. = 0.95420" + MTOL. = 0.95420"

S*E-(.1*P)

Shell

----- External Pressure Calculations -----

Ext Press Chart....= 1 CS-2
Actual Chart Used..= 1 CS-2
Material.....= SA-516-70N
OD.....= 42.50000
T.....= 1.12500
L.....= 236.00000
Temp.....= 725.00000
Stress from MT....= 16450.00000
Yield from MT....= 26750.00000
2 x Stress.....= 32900.00000
B1 from Chart@Right= 0.00000
Yield= 2 x B1 x 0.9= 0.00000
S.....= 0.00000
DO/T.....= 37.77778
L/DO.....= 5.55294
A.....= 0.00095
B.....= 7646.76953
PA1.....= 0.00000
PA2.....= 0.00000
PA.....= 269.88599

Errors/Warnings:

- None

Ellip. Head External pressure calculations:

Ext Press Chart....= 1 CS-2
Actual Chart Used..= 1 CS-2
Material.....= SA-516-70N
OD.....= 42.25000
RO.....= 38.02500
T.....= 0.87500
Temp.....= 725.00000
Stress from MT....= 16450.00000
Yield from MT....= 26750.00000
A.....= 0.00288
B.....= 9705.76465
PA1.....= 223.34108
PA2.....= 423.79901
PA.....= 223.34108

Errors/Warnings:

None

FRONT CHAN.

```

=====
NOM.
SIZE          PRESS.          TEMP.          C.A.          RT          H.T.          TEMA
40.000"      525.0psig        425deg.F      0.1250"      FULL        0deg.F       R
=====

```

ASME CODE SEC.VIII DIV.I UG-27 [CYLINDER]

====DESIGN CONDITIONS=====CYLINDER CALCULATIONS ON I.D. FORMULA====

CHAN. CYL

MATERIAL: SA-516-70 PRESS = 525.0psig TEMP = 425deg.F

```

I.D.          NOM. THK.      RW=0/PIPE=1  CORR.ALLOW  EFFICIENCY
40.0000"      1.0000"        0            0.1250"    1.00

```

```

ATM.STRS      DES.STRS      STRS/STRS    MAWP N-C    MAWP CORR.
20000.0psi    20000.0psi    1.000        970.874psig 847.458psig

```

P*(R+CA)

----- = 0.53673" + CORR. ALLOW. = 0.66173" + MTOL. = 0.66173"

S*E-(.6*P)

Channel

----- External Pressure Calculations -----

Ext Press Chart....=	1	CS-2
Actual Chart Used..=	1	CS-2
Material.....=	SA-516-70	
OD.....=	42.00000	
T.....=	0.87500	
L.....=	24.68750	
Temp.....=	425.00000	
Stress from MT....=	20000.00000	
Yield from MT....=	32125.00000	
2 x Stress.....=	40000.00000	
B1 from Chart@Right=	0.00000	
Yield= 2 x B1 x 0.9=	0.00000	
S.....=	0.00000	
DO/T.....=	48.00000	
L/DO.....=	0.58780	
A.....=	0.00743	
B.....=	15193.60156	
PA1.....=	0.00000	
PA2.....=	0.00000	
PA.....=	422.04449	

Errors/Warnings:

- None

ASME SEC.VIII DIV.1 APPENDIX 2		Front Channel to cover flange	
DESIGN CONDITIONS		GASKET and BOLTING CALCULATIONS TABLE	
Design Pres= 525.0 psig	Eff.Gsk OD= 42.7500"	No.Passes= 2	
Neg. Pres= None	Eff.Gsk ID= 40.0000"	N= 1.375"	
Design Temp= 425 deg.F	THK= 0.1250"	col.=2	b= 0.415
Flg Matl=SA-105	Gsk Matl=GMGC FLEXPLO STYLE PN	y= 2500psi	
Bolt Matl=SA-193B7	Gsk Face=	w= 0.0000"	m= 2.000
Corr Allow= 0.1250"	Wm2= 155247 #	Am = 34.2010in ²	
Flange Desn Sfo 19900.0 psi	Hp = 130408 #	Ab = 34.9440in ²	
Atm. Sfa 20000.0 psi	H = 724617 #	W = 864312 #	
Bolting Desn Sb 25000.0 psi	Wm1= 855025 #	Wm1= 855025 #	
Atm. Sa 25000.0 psi	Gasket Width Check	Nmin = 1.3267"	
CONDITION	LOAD	x	LEVER ARM = MOMENT
	HD = 668006 #		hD= 2.15625" MD= 1440388 in#
Operating	HG = 130408 #		hG= 1.97708" MG= 257827 in#
	HT = 56611 #		hT= 2.39479" MT= 135571 in#
			Mo= 1833788 in#
Gasket			
Seating	HG = 864312 #		hG= 1.97708" mo= 1708813 in#
Allow.Stress-STRESS CALCULATIONS-Operating			SHAPE CONSTANTS
1.5 Sfo&2.5 Sno Long Hub,SH	23418.4 psi	K = 1.1957	h/ho= 0.2738
Sfo Radial Flg,SR	3416.5 psi	T = 1.8407	F = 0.8830
Sfo Tang Flg,ST	9160.5 psi	Z = 5.6557	V = 0.3805
Sfo .5(SH+SR)or.5(SH+ST)	16289.5 psi	Y =10.9627	f = 1.2328
J(APP.2-14)= 0.6959		U =12.0469	e = 0.1488in ⁻¹
Allow.Stress-STRESS CALCULATIONS-Gsk.Seating		g1/go= 1.5000	d = 143.838In ³
1.5 Sfa&2.5 Sna Long Hub,SH	21822.4 psi	ho = 5.9345"	
Sfa Radial Flg,SR	3183.7 psi		
Sfa Tang Flg,ST	8536.2 psi		OTHER STRESS FORMULA FACTORS
Sfa .5(SH+SR)or.5(SH+ST)	15179.3 psi	t	4.1875"
J(APP.2-14)= 0.6118		Alpha	1.6231
		Beta	1.8308
		Gamma	0.8818
		Delta	0.5105
		Lambda	1.3923
		M	45559 #
		m	42454 #
O.D. = 48.1250"	THK. = 4.3750"		
I.D. = 40.0000"	T-Adder = 0.0000"		
GO = 1.0000"	G1 = 1.4375"		
HUB O.D. = 42.8750"	HUB LEN = 1.6250"		
HUB ANG = 15.0685deg	RIB LENGTH = 40.0000"		
G(MEAN) = 41.3750"	G(CALC) = 41.9208"		
G(MIN.) = 41.3750"	G(MAX.) = 42.7500"		
B.C. = 45.8750"	B.S.C.F. = 1.0000		
No. BOLTS = 48	BOLT DIAMETER = 1.1250"		
R = 1.5000"	E = 1.1250"		
BOLT SPAC = 3.0004"	TORQUE = 480ft#		
MIN. SPAC = 2.5000"	TEMA MAX. SPAC= 12.3000"		
FLG TURN = 0.1875"	FACING = RECESS		
BPRIME = 0.0000"	BG1 = 0.0000"		
FLG RWT = 956Lbs	FLG FWT = 737Lbs		

ASME SEC.VIII DIV.1 APPENDIX 2		Front chan to tubesheet flange	
DESIGN CONDITIONS		GASKET and BOLTING CALCULATIONS TABLE	
Design Pres=	525.0 psig	Eff.Gsk OD=	43.7500"
Neg. Pres=	15.0 psig	Eff.Gsk ID=	40.0000"
Design Temp=	425 deg.F	THK=	0.1250"
Flg Matl=	SA-105	Gsk Matl=	GMGC FLEXPLO STYLE PN
Bolt Matl=	SA-193B7	Gsk Face=	w= 0.0000" m= 2.000
Corr Allow=	0.1250"	Wm2=	181418 #
Flange Desn Sfo	19900.0 psi	Hp =	152391 #
Atm. Sfa	20000.0 psi	H =	754685 #
Bolting Desn Sb	24300.0 psi	Wm1=	1145992 #
Atm. Sa	25000.0 psi	Gasket Width Check	Nmin = 1.7971"
CONDITION	LOAD	x	LEVER ARM =
	HD =	668006 #	hD= 2.78125" MD= 1857892 in#
Operating	HG =	391306 #	hG= 2.17162" MG= 849770 in#
	HT =	86679 #	hT= 2.80456" MT= 243096 in#
			Mo= 2950760 in#
Gasket			
Seating	HG =	1193352 #	hG= 2.17162" mo= 2591511 in#
Allow.Stress-STRESS CALCULATIONS-Operating		SHAPE CONSTANTS	
1.5 Sfo&2.5 Sno Long Hub,SH	22856.0 psi	K	= 1.2329 h/ho= 0.2738
Sfo Radial Flg,SR	2245.5 psi	T	= 1.8255 F = 0.8830
Sfo Tang Flg,ST	12410.5 psi	Z	= 4.8455 V = 0.3805
Sfo .5(SH+SR)or.5(SH+ST)	17633.2 psi	Y	= 9.3933 f = 1.2328
J(APP.2-14)=	0.6792	U	=10.3223 e = 0.1488in^-1
Allow.Stress-STRESS CALCULATIONS-Gsk.Seating		g1/go= 1.5000 d = 123.246In^3	
1.5 Sfa&2.5 Sna Long Hub,SH	20073.3 psi	ho	= 5.9345"
Sfa Radial Flg,SR	1972.1 psi	OTHER STRESS FORMULA FACTORS	
Sfa Tang Flg,ST	10899.5 psi	t	5.4375"
Sfa .5(SH+SR)or.5(SH+ST)	15486.4 psi	Alpha	1.8091
J(APP.2-14)=	0.5628	Beta	2.0788
		Gamma	0.9910
		Delta	1.3044
		Lambda	2.2954
		M	73310 #
		m	64385 #
O.D.	= 49.6250"	THK.	= 5.6250"
I.D.	= 40.0000"	T-Adder	= 0.0000"
GO	= 1.0000"	G1	= 1.4375"
HUB O.D.	= 42.8750"	HUB LEN	= 1.6250"
HUB ANG	= 15.0685deg	RIB LENGTH	= 40.0000"
G(MEAN)	= 41.8750"	G(CALC)	= 42.7818"
G(MIN.)	= 41.8750"	G(MAX.)	= 43.3750"
B.C.	= 47.1250"	B.S.C.F.	= 1.0000
No. BOLTS	= 52	BOLT DIAMETER	= 1.2500"
R	= 2.1250"	E	= 1.2500"
BOLT SPAC	= 2.8453"	TORQUE	= 680ft#
MIN. SPAC	= 2.8125"	TEMA MAX. SPAC	= 15.5500"
FLG TURN	= 0.1875"	FACING	= RECESS
BPRIME	= 0.0000"	BG1	= 0.0000"
FLG RWT	= 1393Lbs	FLG FWT	= 1100Lbs

ASME SEC.VIII DIV.1 APPENDIX 2		Shell to tubesheet flange	
DESIGN CONDITIONS		GASKET and BOLTING CALCULATIONS	
Design Pres=	675.0 psig	Eff.Gsk OD=	43.7500"
Neg. Pres=	15.0 psig	Eff.Gsk ID=	40.0000"
Design Temp=	725 deg.F	THK=	0.1250"
Flg Matl=	SA-105/SA-350-LF2	Gsk Matl=	GMGC FLEXPLO STYLE PN
Bolt Matl=	SA-193B7	Gsk Face=	w= 0.0000" m= 2.000
Corr Allow=	0.1250"	Wm2=	181418 #
Flange Desn Sfo	16000.0 psi	Hp =	175682 #
Atm. Sfa	20000.0 psi	H =	970309 #
Bolting Desn Sb	24300.0 psi	Wm1=	1145992 #
Atm. Sa	25000.0 psi	Gasket Width Check	Nmin = 1.7971"
CONDITION	LOAD	x	LEVER ARM =
	HD =	858865 #	hD= 2.59375" MD= 2227682 in#
Operating	HG =	175682 #	hG= 2.17162" MG= 381515 in#
	HT =	111444 #	hT= 2.80456" MT= 312553 in#
			Mo= 2921750 in#
Gasket			
Seating	HG =	1193352 #	hG= 2.17162" mo= 2591511 in#
Allow.Stress--STRESS CALCULATIONS--Operating			SHAPE CONSTANTS
1.5 Sfo&2.5 Sno Long Hub,SH	21901.9 psi	K	= 1.2329 h/ho= 0.2694
Sfo Radial Flg,SR	3752.6 psi	T	= 1.8255 F = 0.8837
Sfo Tang Flg,ST	9091.1 psi	Z	= 4.8455 V = 0.3824
Sfo .5(SH+SR)or.5(SH+ST)	15496.5 psi	Y	= 9.3933 f = 1.2468
J(APP.2-14)=	0.6291	U	=10.3223 e = 0.1313in^-1
Allow.Stress--STRESS CALCULATIONS--Gsk.Seating		g1/go=	1.5000 d = 229.907In^3
1.5 Sfa&2.5 Sna Long Hub,SH	19426.4 psi	ho =	6.7291"
Sfa Radial Flg,SR	3328.5 psi		
Sfa Tang Flg,ST	8063.6 psi		
Sfa .5(SH+SR)or.5(SH+ST)	13745.0 psi		
J(APP.2-14)=	0.4772		
			OTHER STRESS FORMULA FACTORS
		t	5.0000"
		Alpha	1.6566
		Beta	1.8755
		Gamma	0.9075
		Delta	0.5437
		Lambda	1.4512
		M	72590 #
		m	64385 #
O.D.	= 49.6250"	THK.	= 5.1875"
I.D.	= 40.0000"	T-Adder	= 0.0000"
GO	= 1.2500"	G1	= 1.8125"
HUB O.D.	= 43.6250"	HUB LEN	= 1.8125"
HUB ANG	= 17.2415deg	RIB LENGTH	= 0.0000"
G(MEAN)	= 41.8750"	G(CALC)	= 42.7818"
G(MIN.)	= 41.8750"	G(MAX.)	= 43.3750"
B.C.	= 47.1250"	B.S.C.F.	= 1.0000
No. BOLTS	= 52	BOLT DIAMETER	= 1.2500"
R	= 1.7500"	E	= 1.2500"
BOLT SPAC	= 2.8453"	TORQUE	= 680ft#
MIN. SPAC	= 2.8125"	TEMA MAX. SPAC	= 14.5000"
FLG TURN	= 0.1875"	FACING	= RECESS
BPRIME	= 0.0000"	BG1	= 0.0000"
FLG RWT	= 1345Lbs	FLG FWT	= 1049Lbs

MAIN FLG IS FLANGE MARK # 2 Front Channel to cover flange
COMP FLG IS FLANGE MARK # 0

CODE 46 MAIN FLG. SET: GASKET WIDTH WILL NOT CONTROL MAWP.

/// INDIVIDUAL COMPONENTS ///

MAWP CORR. MAIN FLG = 536.0 Limited by: Minimum bolting area Am.
MAWP CORR. COMP FLG = 0.0
MAWP CORR. COVER = 536.0
MAWP CORR. T.S.[STD DIA]TEMA= 0.0

MAWP N-C MAIN FLG = 536.0 Limited by: Minimum bolting area Am.
MAWP N-C COMP FLG = 0.0
MAWP N-C COVER = 536.0
MAWP N-C T.S.[STD DIA]TEMA= 0.0

/// AS MATING FLANGES (NO VACCUUM INCLUDED) ///

MAWP CORR. MAIN FLG = 0.0
MAWP CORR. COMP FLG = 0.0

MAWP N-C MAIN FLG = 0.0
MAWP N-C COMP FLG = 0.0

NOTE: TEMA TUBESHEETS NOT CONSIDERED.

MAIN FLG IS FLANGE MARK # 3 Front chan to tubesheet flange
COMP FLG IS FLANGE MARK # 7 Shell to tubesheet flange

CODE 46 MAIN FLG. SET: GASKET WIDTH WILL NOT CONTROL MAWP.
CODE 46 COMP FLG. SET: GASKET WIDTH WILL NOT CONTROL MAWP.

/// INDIVIDUAL COMPONENTS ///

MAWP CORR. MAIN FLG = 679.0 Limited by: Minimum bolting area Am.
MAWP CORR. COMP FLG = 675.0 Limited by: Minimum bolting area Am.
MAWP CORR. COVER = 0.0
MAWP CORR. T.S.[STD DIA]TEMA= 0.0

MAWP N-C MAIN FLG = 698.0 Limited by: Minimum bolting area Am.
MAWP N-C COMP FLG = 714.0 Limited by: Minimum bolting area Am.
MAWP N-C COVER = 0.0
MAWP N-C T.S.[STD DIA]TEMA= 0.0

/// AS MATING FLANGES (NO VACCUUM INCLUDED) ///

MAWP CORR. MAIN FLG = 679.0 Limited by: Minimum bolting area Am.
MAWP CORR. COMP FLG = 675.0 Limited by: Minimum bolting area Am.

MAWP N-C MAIN FLG = 698.0 Limited by: Minimum bolting area Am.
MAWP N-C COMP FLG = 714.0 Limited by: Minimum bolting area Am.

NOTE: TEMA TUBESHEETS NOT CONSIDERED.

///// FRONT COVER CALCULATIONS /////

EXCHANGER TYPE....= A E U MATERIAL.....= SA-516-70N
 PRESSURE.....= 525.0 psig CORROSION ALLOWANCE= 0.1250"
 TURN LOW.....= 0.1875" TURN HIGH.....= 0.2500"
 S COVER STRESS...= 20000.0 psi E.....= 1.0
 Sb STRESS BOLTS..= 25000.0 psi Ab.....= 34.9440 in^2
 d.....= 41.9208" C.....= 0.3000
 W.....= 864312 # Wm1.....= 855025 #
 hG (ASME).....= 1.9771" hG (TEMA).....= 1.9771"
 G (TEMA).....= 41.9208" E (MOD.) x (10)-6...= 27.6 psi
 COVER THK SET HOLD= [YES]

[PER ASME SECT.VIII UG-34 USING Wm1 (OPERATING)]

$$THK = d \frac{C P}{S E} + \frac{1.9 W m1 hG}{S E d^3} = 4.2036"$$

[PER ASME SECT.VIII UG-34 USING W (GASKET SEATING)]

$$THK = d \frac{1.9 W hG}{S E d^3} = 1.9679"$$

[PER TEMA RCB-9.2 FLAT CHANNEL COVER]

MAXIMUM DEFLECTION = 40.0000" x 0.00125 = 0.0500"
 DEFLECTION at 4.2632" THICK = 0.0500"

$$Y = \frac{G}{E (4.312)^3} | 0.0435 G^3 P + 0.5 Sb Ab hG | = 0.0483"$$

[COV THK Min] TEMA Deflection = 4.2632" + 0.1875" = 4.4507"
 [COV THK Min] CODE Operating = 4.2036" + 0.1875" = 4.3911"
 [COV THK Min] CODE Gasket Seating = 1.9679" + 0.1875" = 2.1554"
 ACTUAL COVER THICKNESS USED (SET HOLD[YES])= 4.5000"

***** TUBE CALCULATIONS *****

Tube size.....=	1.00000"	Straight or U-Tube.=	U-Tube
Internal Press...=	540.00000psig	External press.....=	690.00000psig
Temperature used.=	725.00000deg.	Tube Matl MT# 61...=	SA-214
Stress Oper.....=	9800.00000psi	Stress Atm.....=	11400.00000psi
Yield.....=	18300.00000psi	Inside Radius.....=	0.39100"
Joint Eff.....=	1.00000	Mill tol...= 0.00% =	0.00000"
Wall Thk. t.....=	0.10900"	Tube gauge.....=	12 BWG Gauge
Mean Rad., R.....=	1.50000"	Tube Length.....=	240.00000"
Corrosion ID.....=	0.00000"	Corrosion OD.....=	0.00000"
Thinning factor..=	1.16667		
tr = PR/SE - .6P =	0.02228"	(tr*TF)+MT+CA.....=	0.02600"
MAWP Internal H/C=	2048.06030psig	MAWP Internal N/C..=	2382.43750psig
MAWP External H/C=	1173.06738psig	MAWP External N/C..=	1719.08594psig

/// TEMA PASS PLATE CALCULATIONS ///

Per RCB-9.132

$$t = b \sqrt{\frac{q B}{1.5 S}}$$

TUBE SIDE PRESSURE DROP $q = 15.00000$

FRONT PASS PLATE THK..... = 0.62500

MATERIAL..... = SA-36

STRESS OPER..... = 16600.00

FRONT Chan ID..... = 40.00000

FRONT Chan OAL..... = 42.12500

a..... = 40.00000

b..... = 37.93750

TABLE RCB-9.132..... = Short sides fixed

a/b..... = 1.05437

B..... = 0.44609

t..... = 0.62191

Min fillet leg [both sides] = $3/4t = .75 \times 0.62191 = 0.46643$

---- SHELL NOZZLES -----DESIGN PRESS= 675.0 DESIGN TEMP= 725.0
NOZ#S1 12.0- 600# MT# FLG= 70 GROUP=1.1 @70deg.=1480.00 @TEMP=1037.50
NOZ#S2 6.0- 600# MT# FLG= 70 GROUP=1.1 @70deg.=1480.00 @TEMP=1037.50

---- FRONT CHANNEL NOZZLES -----DESIGN PRESS= 525.0 DESIGN TEMP= 425.0
NOZ#T1 10.0- 300# MT# FLG= 70 GROUP=1.1 @70deg.= 740.00 @TEMP= 627.50
NOZ#T2 14.0- 300# MT# FLG= 70 GROUP=1.1 @70deg.= 740.00 @TEMP= 627.50

---- REAR CHANNEL/COVER NOZZLES ----DESIGN PRESS= 525.0 DESIGN TEMP= 425.0

NOZZLE REINFORCEMENT CALCULATIONS/ ASME SEC.VIII DIV.I UG-37 & UG-16

NOZ NO=S1 12.00" 600 LB RF WN
[INLET] SHELL SIDE
PRESSURE = 675.0psig TEMP = 725.0deg.F

-----CYLINDER-----NOZZLE-----

SA-516-70N		SA-106B	
OPER STRS=	16450.0 psi	OPER STRS=	14300.0 psi
ATM STRS =	20000.0 psi	ATM STRS =	17100.0 psi
C.A. =	0.12500"	C.A. =	0.12500"
I.D. =	40.00000"	I.D. =	11.37400"
O.D. =	42.50000"	O.D. =	12.75000"
THK. =	1.25000"	THK. sc80=	0.68800"
T =	1.12500"	TN =	0.56300"
TR =	0.84664"	TRN =	0.29534"
ET =	0.27836"	ETN =	0.26766"

MIN STD. WT. PIPE+C.A.= 0.45312" TR+C.A.= 0.97164"
LIMITS= 23.24800"
fr1= 0.8693 fr2= 0.8693 fr3= 0.8693 fr4= 1.0000 F= 1.00000
Min weld sizes: NS= 0.0000" NP= 0.3536" PS= 0.5303" IF= 0.0000"

AREA REQD. A = 9.96597in^2
EXCESS IN CYL. A1= 3.19467in^2
EXCESS IN NOZ.(OUTSIDE) A2= 1.12033in^2 H = 2.40750"
AREA OF FILLETS = 0.68475in^2
AREA AVAILABLE = 4.99974in^2

NOZ. NECK WELD= 0.37500" AREA= 0.12225in^2

PAD MATL.= SA-516-70N S= 16450.psi
PAD O.D.= 18.50000" THK.= 1.00000" AREA= 5.75000in^2
PAD TO CYL. WELD= 0.75000" AREA= 0.56250in^2
ARC= 12.949"
PAD RET. DIM.= 18.500" X 18.750"

< AREA REQD-AREA AVAILABLE > = -0.78378in^2

MAWP N-C= 940.0psig LIMITED BY AR
MAWP COR= 699.0psig LIMITED BY AR

All Weld Paths are stronger than the required strength /[UG-41 UG-45 UW-15]

NOZZLE REINFORCEMENT CALCULATIONS/ ASME SEC.VIII DIV.I UG-37 & UG-16

NOZ NO=S2 6.00" 600 LB RF WN

[OUTLET] SHELL SIDE

PRESSURE = 675.0psig TEMP = 725.0deg.F

-----CYLINDER-----NOZZLE-----

SA-516-70N		SA-106B	
OPER STRS=	16450.0 psi	OPER STRS=	14300.0 psi
ATM STRS =	20000.0 psi	ATM STRS =	17100.0 psi
C.A. =	0.12500"	C.A. =	0.12500"
I.D. =	40.00000"	I.D. =	5.76100"
O.D. =	42.50000"	O.D. =	6.62500"
THK. =	1.25000"	THK. XH=	0.43200"
T =	1.12500"	TN =	0.30700"
TR =	0.84664"	TRN =	0.15346"
ET =	0.27836"	ETN =	0.15354"

MIN STD. WT. PIPE+C.A.= 0.37000" TR+C.A.= 0.97164"

LIMITS= 12.02200"

fr1= 0.8693 fr2= 0.8693 fr3= 0.8693 fr4= 1.0000 F= 1.00000

Min weld sizes:NS= 0.0000" NP= 0.3039" PS= 0.5303" IF= 0.0000"

AREA REQD.	A =	5.15711in^2	
EXCESS IN CYL.	A1=	1.65087in^2	
EXCESS IN NOZ.(OUTSIDE)	A2=	0.43845in^2	H = 1.64250"
AREA OF FILLETS	=	0.59490in^2	
AREA AVAILABLE	=	2.68422in^2	

NOZ. NECK WELD= 0.37500" AREA= 0.12225in^2

PAD MATL.= SA-516-70N	S=	16450.psi	
PAD O.D.= 9.62500"	THK.=	0.87500"	AREA= 2.62500in^2
PAD TO CYL. WELD= 0.68750"	AREA=	0.47266in^2	
ARC= 6.652"			
PAD RET. DIM.= 9.625" X		9.688"	

< AREA REQD-AREA AVAILABLE > = -0.15211in^2

MAWP N-C= 937.0psig LIMITED BY AR

MAWP COR= 684.0psig LIMITED BY AR

All Weld Paths are stronger than the required strength /[UG-41 UG-45 UW-15]

NOZZLE REINFORCEMENT CALCULATIONS/ ASME SEC.VIII DIV.I UG-37 & UG-16

NOZ NO= S3 1.00" 600 LB RF WN

[VENT] SHELL SIDE

PRESSURE = 675.0psig TEMP = 725.0deg.F

-----CYLINDER-----NOZZLE-----

SA-516-70N

SA-106B

OPER STRS= 16450.0 psi

OPER STRS= 14300.0 psi

ATM STRS = 20000.0 psi

ATM STRS = 17100.0 psi

C.A. = 0.12500"

C.A. = 0.12500"

I.D. = 40.00000"

I.D. = 0.59900"

O.D. = 42.50000"

O.D. = 1.31500"

THK. = 1.25000"

THK. = 0.35800"

T = 1.12500"

TN = 0.23300"

TR = 0.84664"

TRN = 0.03046"

ET = 0.27836"

ETN = 0.20254"

MIN STD. WT. PIPE+C.A.= 0.24137" TR+C.A.= 0.97164"

LIMITS= 3.56500"

fr1= 0.8693 fr2= 0.8693 fr3= 0.0000 fr4= 0.0000 F= 1.00000

Min weld sizes:NS= 0.2307" NP= 0.0000" PS= 0.0000" IF= 0.0000"

NOZ. NECK WELD= 0.37500"

MAWP N-C= 1137.0psig LIMITED BY AR

MAWP COR= 738.0psig LIMITED BY AR

Note: Taking a UG-36(c)(3)(a) exemption for nozzle S3.

This calculation is valid for nozzles that meet all the requirements of paragraph UG-36.

NOZZLE REINFORCEMENT CALCULATIONS/ ASME SEC.VIII DIV.I UG-37 & UG-16

NOZ NO=T1 10.00" 300 LB RF WN

[INLET] FRONT CHANNEL SIDE

PRESSURE = 525.0psig TEMP = 425.0deg.F

-----CYLINDER-----NOZZLE-----

SA-516-70		SA-106B	
OPER STRS=	20000.0 psi	OPER STRS=	17100.0 psi
ATM STRS =	20000.0 psi	ATM STRS =	17100.0 psi
C.A.	= 0.12500"	C.A.	= 0.12500"
I.D.	= 40.00000"	I.D.	= 9.06200"
O.D.	= 42.00000"	O.D.	= 10.75000"
THK.	= 1.00000"	THK.sc120=	0.84400"
T	= 0.87500"	TN	= 0.71900"
TR	= 0.53673"	TRN	= 0.16302"
ET	= 0.33827"	ETN	= 0.55598"

MIN STD. WT. PIPE+C.A.= 0.44438" TR+C.A.= 0.66173"

LIMITS= 18.62400"

fr1= 0.8550 fr2= 0.8550 fr3= 0.0000 fr4= 0.0000 F= 1.00000

Min weld sizes:NS= 0.3536" NP= 0.0000" PS= 0.0000" IF= 0.0000"

AREA REQD.	A =	5.10999in^2	
EXCESS IN CYL.	A1=	3.07939in^2	
EXCESS IN NOZ.(OUTSIDE)	A2=	1.70893in^2	H = 1.79750"
AREA OF FILLETS	=	0.85500in^2	
AREA AVAILABLE	=	5.64332in^2	

NOZ. NECK WELD= 1.00000" AREA= 0.85500in^2

< AREA REQD-AREA AVAILABLE > = -0.53333in^2

MAWP N-C= 648.0psig LIMITED BY AR

MAWP COR= 551.0psig LIMITED BY AR

NOZZLE ATTACHMENT WELD LOADS AND WELD STRENGTH PATHS: UG41 UG45 UW15

NOZ NO=T1 10.00" 300 LB RF WN

FR1 = 0.85500 S_SHL = 20000.00 S_NOZ = 17100.00 A = 5.33944
 A1 = 4.34680 T_NOZ = 0.59400 C_NOZ = 0.12500 OD_NOZ= 10.75000
 T_SHL = 0.75000 Tmin = 0.53673 C_SHL = 0.12500 F = 1.00000
 A2 = 0.81754 A5 = 3.65625 A41 = 0.12023 A42 = 0.31641
 A43 = 0.00000 E1 = 1.00000 PAD TK= 0.75000

----- Load to be carried by welds per UG41(b)(2) -----
 W = [A - A1 + 2 tn fr1 (E1 t - F tr)] Sv = 91123.57

----- Load to be carried by welds per UG41(b)(1) -----
 W1-1 = (A2 + A5 + A41 + A42) Sv = 98208.62
 W2-2 = (A2 + A3 + A41 + A43 + 2 tn t fr1) Sv = 28780.37
 W3-3 = (A2 + A3 + A5 + A41 + A42 + A43 + 2 tn t fr1) Sv = 108233.50

----- Unit stresses -----
 Groove weld pad to nozzle = 12654.00 Fillet at pad OD = 9800.00
 Ext. fillet at nozzle = 8379.00 Groove weld comp. to nozzle= 12654.00
 Nozzle wall shear = 11970.00 Int. fillet at nozzle = 8379.00

----- Strength of connection elements -----
 Groove weld pad to nozzle = 160257.03 Fillet at pad OD = 135297.00
 Ext. fillet at nozzle = 53058.07 Groove weld comp. to nozzle= 133547.52
 Nozzle wall shear = 90661.28 Int. fillet at nozzle = 0.00

----- Check Strength Paths -----
 Str. thru path 1-1 must be >= the lesser of W or W1-1 per UG-41(b)
 Str. thru path 2-2 must be >= the lesser of W or W2-2 per UG-41(b)
 Str. thru path 3-3 must be >= the lesser of W or W3-3 per UG-41(b)
 Str. thru path 1-1 (90661.3 + 135297.0) = 225958.28
 Str. thru path 2-2 (53058.1 + 133547.5 + 160257.0 + 0.0) = 346862.62
 Str. thru path 3-3 (133547.5 + 135297.0) = 268844.50

NOZZLE REINFORCEMENT CALCULATIONS/ ASME SEC.VIII DIV.I UG-37 & UG-16

NOZ NO=T2 14.00" 300 LB RF WN

[OUTLET] FRONT CHANNEL SIDE

PRESSURE = 525.0psig TEMP = 425.0deg.F

-----CYLINDER-----NOZZLE-----

SA-516-70		SA-106B	
OPER STRS=	20000.0 psi	OPER STRS=	17100.0 psi
ATM STRS =	20000.0 psi	ATM STRS =	17100.0 psi
C.A.	= 0.12500"	C.A.	= 0.12500"
I.D.	= 40.00000"	I.D.	= 12.12400"
O.D.	= 42.00000"	O.D.	= 14.00000"
THK.	= 1.00000"	THK.sc100=	0.93800"
T	= 0.87500"	TN	= 0.81300"
TR	= 0.53673"	TRN	= 0.21231"
ET	= 0.33827"	ETN	= 0.60070"

MIN STD. WT. PIPE+C.A.= 0.45312" TR+C.A.= 0.66173"

LIMITS= 24.74800"

fr1= 0.8550 fr2= 0.8550 fr3= 0.0000 fr4= 0.0000 F= 1.00000

Min weld sizes:NS= 0.3536" NP= 0.0000" PS= 0.0000" IF= 0.0000"

AREA REQD.	A =	6.76810in^2	
EXCESS IN CYL.	A1=	4.10594in^2	
EXCESS IN NOZ.(OUTSIDE)	A2=	2.08776in^2	H = 2.03250"
AREA OF FILLETS	=	0.85500in^2	
AREA AVAILABLE	=	7.04870in^2	

NOZ. NECK WELD= 1.00000" AREA= 0.85500in^2

< AREA REQD-AREA AVAILABLE > = -0.28060in^2

MAWP N-C= 625.0psig LIMITED BY AR

MAWP COR= 535.0psig LIMITED BY AR

NOZZLE ATTACHMENT WELD LOADS AND WELD STRENGTH PATHS: UG41 UG45 UW15

NOZ NO=T2 14.00" 300 LB RF WN

FR1 = 0.85500 S_SHL = 20000.00 S_NOZ = 17100.00 A = 6.94065
 A1 = 5.62934 T_NOZ = 0.75000 C_NOZ = 0.12500 OD_NOZ= 14.00000
 T_SHL = 0.75000 Tmin = 0.53673 C_SHL = 0.12500 F = 1.00000
 A2 = 1.10267 A5 = 4.78125 A41 = 0.12023 A42 = 0.31641
 A43 = 0.00000 E1 = 1.00000 PAD TK= 0.75000

----- Load to be carried by welds per UG41(b)(2) -----
 W = [A - A1 + 2 tn fr1 (E1 t - F tr)] Sv = 118512.05

----- Load to be carried by welds per UG41(b)(1) -----
 W1-1 = (A2 + A5 + A41 + A42) Sv = 126411.20
 W2-2 = (A2 + A3 + A41 + A43 + 2 tn t fr1) Sv = 37817.45
 W3-3 = (A2 + A3 + A5 + A41 + A42 + A43 + 2 tn t fr1) Sv = 139770.58

----- Unit stresses -----
 Groove weld pad to nozzle = 12654.00 Fillet at pad OD = 9800.00
 Ext. fillet at nozzle = 8379.00 Groove weld comp. to nozzle= 12654.00
 Nozzle wall shear = 11970.00 Int. fillet at nozzle = 8379.00

----- Strength of connection elements -----
 Groove weld pad to nozzle = 208706.83 Fillet at pad OD = 176427.28
 Ext. fillet at nozzle = 69098.88 Groove weld comp. to nozzle= 173922.36
 Nozzle wall shear = 157176.45 Int. fillet at nozzle = 0.00

----- Check Strength Paths -----
 Str. thru path 1-1 must be >= the lesser of W or W1-1 per UG-41(b)
 Str. thru path 2-2 must be >= the lesser of W or W2-2 per UG-41(b)
 Str. thru path 3-3 must be >= the lesser of W or W3-3 per UG-41(b)
 Str. thru path 1-1 (157176.5 + 176427.3) = 333603.75
 Str. thru path 2-2 (69098.9 + 173922.4 + 208706.8 + 0.0) = 451728.06
 Str. thru path 3-3 (173922.4 + 176427.3) = 350349.62

(U-Tube) Configuration d

No or EPC=Elastic-Plastic Calcs	SSC=Simply Supported Calcs	=NO
MT Tubes	=SA-214	MT Tubesheets =SA-350LF2CL1
MT Channel	=SA-516-70	MT Shell =SA-516-70N
Diameter (Do)	= 39.5000"	Pass Lane Area (AL) = 139.654in ²
Perimeter of layout (CP)	= 124.0928"	Area enclosed by CP (AP) = 1225.417in ²
Tubesheet thickness (h)	= 6.2500"	Tubesheet thickness HOLD = YES
Tube Pitch (p)	= 1.2500"	Pitch type = Rot Square (45 Deg)
Diameter of tube (dt)	= 1.0000"	Tube thickness (tt) = 0.1090"
Exp length of tube (ltx)	= 6.0000"	Exp depth ratio (Rho) = 0.9792
Groove channel side (hg)	= 0.1875"	Groove shell side = 0.0000"
Channel I.D. (Dc)	= 40.0000"	Channel thickness (Tc) = 1.0000"
Shell I.D. (Ds)	= 40.0000"	Shell thickness (Ts) = 1.2500"
Stat TS diameter (A)	= 39.8750"	Bolt circle (C) = 47.1250"
G Channel side (Gc or G1)	= 42.7817"	G Shell side (Gs or G1) = 42.7817"
Bolt load Chan side (Wc)	=1193352.1#	Bolt load Sh side (Ws) =1193352.1#
1.8* $\sqrt{Dc*Tc}$	= 0.0000"	1.8* $\sqrt{Ds*(Ts \text{ or } Ts1)}$ = 0.0000"
Thickness at Top TS (tr)	= 5.1250"	Outer Tube Circle = 39.5000"
hr Ch Wm1	=1145992.#	hr Ch W =1193352.1#
hr Sh Wm1	=1145992.#	hr Sh W =1193352.1#
hr Ch Op= 2.6283" Ch Atm= 2.3989"		Sh Op= 2.6283" Sh Atm = 2.3989"
DE Stationary Tubesheet = 0.0"		DE Floating Tubesheet = 0.0"
Tubes welded Backside TS =NO		Hole Size in Tubesheet = 0.0000"
Tube Joint= Full Strength Welds Per FIG. UW-20.1(c)		
Design Press Chan (Pt) = 525.0psig	Design Press Shell (Ps) = 675.0psig	
Design Temp Chan = 425.0deg.	Design Temp Shell = 725.0deg.	
Corr Chan = 0.12500"	Corr Shell = 0.12500"	
Corr Chan side TS (ct) = 0.12500"	Corr Shell side TS (cs) = 0.12500"	

(U-Tube) Configuration d

Tubesheet Thickness (h)	=	6.0000"			
Exp length of tube (ltx)	=	5.8750"	Exp depth ratio (Rho)	=	0.9792"
Channel I.D. (Dc)	=	40.2500"	Channel thickness (Tc)	=	0.8750"
Shell I.D. (Ds)	=	40.2500"	Shell thickness (Ts)	=	1.1250"
Perimeter of layout (CP)	=	124.0928"	Area enclosed by CP (AP)	=	1225.417in ²
Press Chan (Pt)	=	525.0psi	Press Shell (Ps)	=	-15.0psi
Corr Chan	=	0.12500"	Corr Shell	=	0.12500"
Corr Chan side TS (ct)	=	0.12500"	Corr Shell side TS (cs)	=	0.12500"

Temp	Stress	Yield	Modulus	Coef	Poisson	Prim Plus
			x10 ⁶	x10 ⁻⁶	Ratio	Stress
Temp TS T = 725deg.F	S = 16450psi	Sy = 0psi	E = 24.9750psi	V = .000	Sps =	0psi
Temp TBS Tt= 725deg.F	St = 11529psi	Syt= 0psi	Et = 24.9750psi	Vt=.000		
Temp TBS T = 725deg.F	St = 11529psi		Et = 24.9750psi			
Temp CH Tc= 425deg.F	Sc = 20000psi	Syc= 32125psi	Ec = 27.5500psi	Vc=.300	SpSC=	64250psi
Temp SH Ts= 725deg.F	Ss = 16450psi	Sys= 26750psi	Es = 24.9750psi	Vs=.300	SpSS=	53500psi

Mu=0.2000	d*=0.8504"	P*=1.32796"	Mu*=0.3596	hg'=0.0625"	hgs'= 0.0"	hrDE= 0.0"
W*	= 1145992.000#	Rho s	= 1.083082318			
Rho c	= 1.083082318	Mts	= -9507.136719In-Lb/In	h/p	= 4.800000191	
E*	= 10252286.00psi	v*	= 0.320897967	Beta s	= 0.0in ⁻¹	
Kappa s	= 0.0#	Lambda s	= 0.0psi	Delta s	= 0.0In ³ /Lb	
Delta c	= 0.0In ³ /Lb	Beta c	= 0.0in ⁻¹	Kappa c	= 0.0#	
Lambda c	= 0.0psi	Omega s	= 0.0in ²	Omega c	= 0.0in ²	
K	= 1.009493709	F	= 0.015631560	M*	= -9507.136719In-Lb/In	
Mp	= -8955.580078In-Lb/In	Mo	= -52673.90625In-Lb/In	M	= 52673.90625In-Lb/In	
Sigma s,m=	0.0psi	Sigma s,b=	0.0psi	Sigma s	= 0.0psi	
Sigma c,m=	0.0psi	Sigma c,b=	0.0psi	Sigma c	= 0.0psi	
Sigma	24928 psi	<= 2S	32900 psi	[OK]	
Tau	4443 psi	<= .8S	13160 psi	[OK]	

Tube Joint= Full Strength Welds Per FIG. UW-20.1(c)

Tube O.D. do=1.000000"
 Tube Thickness t=0.109000"
 Stress Tubes Sa=11529.41psi
 St = Stress Tubesheet or Base if Clad=16450.00psi
 Sw = Stress Weld = Lesser of Sa or St=11529.41psi
 fw = Weld strength factor = Sa/Sw=1.000000
 af = Fillet Weld leg=0.120000"
 ag = Groove Weld leg=0.120000"
 ac = Length of the combined weld legs = af + ag=0.240000"
 Weld throat=0.084853"
 Ft = Axial tube strength = $\Phi \cdot t \cdot (do-t) \cdot Sa$ =3517.719#
 Fd = Design strength, but not greater than Ft=3517.719#
 fd = Ratio of design strength to tube strength
 fd = 1.0 for Full strength welds
 fd = Fd/Ft for Partial strength welds=1.000000
 Fg = Groove weld strength, but not > Ft, = $0.85 \cdot \Phi \cdot ag \cdot (do+0.67 \cdot ag) \cdot Sw$ =3517.719#
 Ff = Fillet weld strength, but not > Ft, = $0.55 \cdot \Phi \cdot af \cdot (do+0.67 \cdot af) \cdot Sw$ =2582.769#
 ff = Ratio of fillet weld strength to design strength = $1.0 - Fg / (fd \cdot Ft)$ = 0.0
 ar = $2.0 \cdot [\text{Sqrt}((0.75 \cdot do)^2 + 1.07 \cdot t \cdot (do-t) \cdot fw \cdot fd) - 0.75 \cdot do]$ =0.132688"
 Lmax = Ft=3517.719#

--- Weld Size Check ---

ac shall not be less than the greater of [ar or t] 0.240" >= 0.133"

(U-Tube) Configuration d

Tubesheet Thickness (h) = 6.0000"
 Exp length of tube (ltx) = 5.8750" Exp depth ratio (Rho) = 0.9792"
 Channel I.D. (Dc) = 40.2500" Channel thickness (Tc) = 0.8750"
 Shell I.D. (Ds) = 40.2500" Shell thickness (Ts) = 1.1250"
 Perimeter of layout (CP) = 124.0928" Area enclosed by CP (AP) = 1225.417in²
 Press Chan (Pt) = -15.0psi Press Shell (Ps) = 675.0psi
 Corr Chan = 0.12500" Corr Shell = 0.12500"
 Corr Chan side TS (ct) = 0.12500" Corr Shell side TS (cs) = 0.12500"

Temp	Stress	Yield	Modulus	Coef	Poisson	Prim Plus
			x10 ⁶	x10 ⁻⁶	Ratio	Stress
Temp TS T = 725deg.F	S = 16450psi	Sy = 0psi	E = 24.9750psi	V = .000	Sps =	0psi
Temp TBS Tt= 725deg.F	St = 11529psi	Syt= 0psi	Et = 24.9750psi	Vt=.000		
Temp TBS T = 725deg.F	St = 11529psi		Et = 24.9750psi			
Temp CH Tc= 425deg.F	Sc = 20000psi	Syc= 32125psi	Ec = 27.5500psi	Vc=.300	SpSC=	64250psi
Temp SH Ts= 725deg.F	Ss = 16450psi	Sys= 26750psi	Es = 24.9750psi	Vs=.300	SpSS=	53500psi

Mu=0.2000 d*=0.8504" P*=1.32796" Mu*=0.3596 hg'=0.0625" hgs'= 0.0" hrDE= 0.0"
 W* = 1145992.000# Rho s = 1.083082318
 Rho c = 1.083082318 Mts = 12148.00879In-Lb/In h/p = 4.800000191
 E* = 10252286.00psi v* = 0.320897967 Beta s = 0.0in⁻¹
 Kappa s = 0.0# Lambda s = 0.0psi Delta s = 0.0In³/Lb
 Delta c = 0.0In³/Lb Beta c = 0.0in⁻¹ Kappa c = 0.0#
 Lambda c = 0.0psi Omega s = 0.0in² Omega c = 0.0in²
 K = 1.009493709 F = 0.015631560 M* = 12148.00879In-Lb/In
 Mp = 11443.24219In-Lb/In Mo = 67305.54688In-Lb/In M = 67305.54688In-Lb/In
 Sigma s,m= 0.0psi Sigma s,b= 0.0psi Sigma s = 0.0psi
 Sigma c,m= 0.0psi Sigma c,b= 0.0psi Sigma c = 0.0psi
 Sigma | 31852 psi | <= 2S 32900 psi [OK]
 Tau | 5678 psi | <= .8S 13160 psi [OK]

Tube Joint= Full Strength Welds Per FIG. UW-20.1(c)

Tube O.D. do=1.000000"
 Tube Thickness t=0.109000"
 Stress Tubes Sa=11529.41psi
 St = Stress Tubesheet or Base if Clad=16450.00psi
 Sw = Stress Weld = Lesser of Sa or St=11529.41psi
 fw = Weld strength factor = Sa/Sw=1.000000
 af = Fillet Weld leg=0.120000"
 ag = Groove Weld leg=0.120000"
 ac = Length of the combined weld legs = af + ag=0.240000"
 Weld throat=0.084853"
 Ft = Axial tube strength = Phi*t*(do-t)*Sa=3517.719#
 Fd = Design strength, but not greater than Ft=3517.719#
 fd = Ratio of design strength to tube strength
 fd = 1.0 for Full strength welds
 fd = Fd/Ft for Partial strength welds=1.000000
 Fg = Groove weld strength, but not > Ft, = 0.85*Phi*ag*(do+0.67*ag)Sw=3517.719#
 Ff = Fillet weld strength, but not > Ft, = 0.55*Phi*af*(do+0.67*af)Sw=2582.769#
 ff = Ratio of fillet weld strength to design strength = 1.0-Fg/(fd*Ft)= 0.0
 ar = 2.0*[Sqrt((0.75*do)^2+1.07*t*(do-t)fw*fd) -0.75*do]=0.132688"
 Lmax = Ft=3517.719#

--- Weld Size Check ---

ac shall not be less than the greater of [ar or t] 0.240" >= 0.133"

(U-Tube) Configuration d

Tubesheet Thickness (h)	=	6.0000"			
Exp length of tube (ltx)	=	5.8750"	Exp depth ratio (Rho)	=	0.9792"
Channel I.D. (Dc)	=	40.2500"	Channel thickness (Tc)	=	0.8750"
Shell I.D. (Ds)	=	40.2500"	Shell thickness (Ts)	=	1.1250"
Perimeter of layout (CP)	=	124.0928"	Area enclosed by CP (AP)	=	1225.417in ²
Press Chan (Pt)	=	525.0psi	Press Shell (Ps)	=	675.0psi
Corr Chan	=	0.12500"	Corr Shell	=	0.12500"
Corr Chan side TS (ct)	=	0.12500"	Corr Shell side TS (cs)	=	0.12500"

Temp	Stress	Yield	Modulus	Coef	Poisson	Prim Plus
			x10 ⁶	x10 ⁻⁶	Ratio	Stress
Temp TS T = 725deg.F	S = 16450psi	Sy = 0psi	E = 24.9750psi	V = .000	Sps =	0psi
Temp TBS Tt= 725deg.F	St = 11529psi	Syt= 0psi	Et = 24.9750psi	Vt=.000		
Temp TBS T = 725deg.F	St = 11529psi		Et = 24.9750psi			
Temp CH Tc= 425deg.F	Sc = 20000psi	Syc= 32125psi	Ec = 27.5500psi	Vc=.300	SpSC=	64250psi
Temp SH Ts= 725deg.F	Ss = 16450psi	Sys= 26750psi	Es = 24.9750psi	Vs=.300	SpSS=	53500psi

Mu=0.2000	d*=0.8504"	P*=1.32796"	Mu*=0.3596	hg'=0.0625"	hgs'= 0.0"	hrDE= 0.0"
W*	= 1145992.000#	Rho s	= 1.083082318			
Rho c	= 1.083082318	Mts	= 2640.871338In-Lb/In	h/p	= 4.800000191	
E*	= 10252286.00psi	v*	= 0.320897967	Beta s	= 0.0in ⁻¹	
Kappa s	= 0.0#	Lambda s	= 0.0psi	Delta s	= 0.0In ³ /Lb	
Delta c	= 0.0In ³ /Lb	Beta c	= 0.0in ⁻¹	Kappa c	= 0.0#	
Lambda c	= 0.0psi	Omega s	= 0.0in ²	Omega c	= 0.0in ²	
K	= 1.009493709	F	= 0.015631560	M*	= 2640.871338In-Lb/In	
Mp	= 2487.661133In-Lb/In	Mo	= 14631.64062In-Lb/In	M	= 14631.64062In-Lb/In	
Sigma s,m=	0.0psi	Sigma s,b=	0.0psi	Sigma s	= 0.0psi	
Sigma c,m=	0.0psi	Sigma c,b=	0.0psi	Sigma c	= 0.0psi	
Sigma	6924 psi	<= 2S	32900 psi	[OK]	
Tau	1234 psi	<= .8S	13160 psi	[OK]	

Tube Joint= Full Strength Welds Per FIG. UW-20.1(c)

Tube O.D. do=1.000000"
 Tube Thickness t=0.109000"
 Stress Tubes Sa=11529.41psi
 St = Stress Tubesheet or Base if Clad=16450.00psi
 Sw = Stress Weld = Lesser of Sa or St=11529.41psi
 fw = Weld strength factor = Sa/Sw=1.000000
 af = Fillet Weld leg=0.120000"
 ag = Groove Weld leg=0.120000"
 ac = Length of the combined weld legs = af + ag=0.240000"
 Weld throat=0.084853"
 Ft = Axial tube strength = $\Phi \cdot t \cdot (do-t) \cdot Sa$ =3517.719#
 Fd = Design strength, but not greater than Ft=3517.719#
 fd = Ratio of design strength to tube strength
 fd = 1.0 for Full strength welds
 fd = Fd/Ft for Partial strength welds=1.000000
 Fg = Groove weld strength, but not > Ft, = $0.85 \cdot \Phi \cdot ag \cdot (do+0.67 \cdot ag) \cdot Sw$ =3517.719#
 Ff = Fillet weld strength, but not > Ft, = $0.55 \cdot \Phi \cdot af \cdot (do+0.67 \cdot af) \cdot Sw$ =2582.769#
 ff = Ratio of fillet weld strength to design strength = $1.0 - Fg / (fd \cdot Ft)$ = 0.0
 ar = $2.0 \cdot [\text{Sqrt}((0.75 \cdot do)^2 + 1.07 \cdot t \cdot (do-t) \cdot fw \cdot fd) - 0.75 \cdot do]$ =0.132688"
 Lmax = Ft=3517.719#

--- Weld Size Check ---

ac shall not be less than the greater of [ar or t] 0.240" >= 0.133"

(U-Tube) Configuration d

Tubesheet Thickness (h) = 6.0000"
 Exp length of tube (ltx) = 5.8750" Exp depth ratio (Rho) = 0.9792"
 Channel I.D. (Dc) = 40.2500" Channel thickness (Tc) = 0.8750"
 Shell I.D. (Ds) = 40.2500" Shell thickness (Ts) = 1.1250"
 Perimeter of layout (CP) = 124.0928" Area enclosed by CP (AP) = 1225.417in²
 Press Chan (Pt) = -15.0psi Press Shell (Ps) = -15.0psi
 Corr Chan = 0.12500" Corr Shell = 0.12500"
 Corr Chan side TS (ct) = 0.12500" Corr Shell side TS (cs) = 0.12500"

Temp	Stress	Yield	Modulus	Coef	Poisson	Prim Plus
			x10 ⁶	x10 ⁻⁶	Ratio	Stress
Temp TS T = 725deg.F	S = 16450psi	Sy = 0psi	E = 24.9750psi	V = .000	Sps =	0psi
Temp TBS Tt= 725deg.F	St = 11529psi	Syt= 0psi	Et = 24.9750psi	Vt=.000		
Temp TBS T = 725deg.F	St = 11529psi		Et = 24.9750psi			
Temp CH Tc= 425deg.F	Sc = 20000psi	Syc= 32125psi	Ec = 27.5500psi	Vc=.300	SpSC=	64250psi
Temp SH Ts= 725deg.F	Ss = 16450psi	Sys= 26750psi	Es = 24.9750psi	Vs=.300	SpSS=	53500psi

Mu=0.2000 d*=0.8504" P*=1.32796" Mu*=0.3596 hg'=0.0625" hgs'= 0.0" hrDE= 0.0"
 W* = 1145992.000# Rho s = 1.083082318
 Rho c = 1.083082318 Mts = 0.0In-Lb/In h/p = 4.800000191
 E* = 10252286.00psi v* = 0.320897967 Beta s = 0.0in⁻¹
 Kappa s = 0.0# Lambda s = 0.0psi Delta s = 0.0In³/Lb
 Delta c = 0.0In³/Lb Beta c = 0.0in⁻¹ Kappa c = 0.0#
 Lambda c = 0.0psi Omega s = 0.0in² Omega c = 0.0in²
 K = 1.009493709 F = 0.015631560 M* = 0.0In-Lb/In
 Mp = 0.0In-Lb/In Mo = 0.0In-Lb/In M = 0.0In-Lb/In
 Sigma s,m= 0.0psi Sigma s,b= 0.0psi Sigma s = 0.0psi
 Sigma c,m= 0.0psi Sigma c,b= 0.0psi Sigma c = 0.0psi
 Sigma | 0 psi | <= 2S 32900 psi [OK]
 Tau | 0 psi | <= .8S 13160 psi [OK]

Tube Joint= Full Strength Welds Per FIG. UW-20.1(c)

Tube O.D. do=1.000000"
 Tube Thickness t=0.109000"
 Stress Tubes Sa=11529.41psi
 St = Stress Tubesheet or Base if Clad=16450.00psi
 Sw = Stress Weld = Lesser of Sa or St=11529.41psi
 fw = Weld strength factor = Sa/Sw=1.000000
 af = Fillet Weld leg=0.120000"
 ag = Groove Weld leg=0.120000"
 ac = Length of the combined weld legs = af + ag=0.240000"
 Weld throat=0.084853"
 Ft = Axial tube strength = Phi*t*(do-t)*Sa=3517.719#
 Fd = Design strength, but not greater than Ft=3517.719#
 fd = Ratio of design strength to tube strength
 fd = 1.0 for Full strength welds
 fd = Fd/Ft for Partial strength welds=1.000000
 Fg = Groove weld strength, but not > Ft, = 0.85*Phi*ag*(do+0.67*ag)Sw=3517.719#
 Ff = Fillet weld strength, but not > Ft, = 0.55*Phi*af*(do+0.67*af)Sw=2582.769#
 ff = Ratio of fillet weld strength to design strength = 1.0-Fg/(fd*Ft)= 0.0
 ar = 2.0*[Sqrt((0.75*do)^2+1.07*t*(do-t)fw*fd) -0.75*do]=0.132688"
 Lmax = Ft=3517.719#

--- Weld Size Check ---

ac shall not be less than the greater of [ar or t] 0.240" >= 0.133"

```

-----
          [  Pressure  ] [ Temperature ]
[Case]   [ Pt   ] [ Ps   ] [T_ch ] [T_sh ] [Corr] [ Thermal Expansion ]
-----
  1      525.0   -15.0   425.0  725.0   Yes    -
  2      -15.0   675.0   425.0  725.0   Yes    -
  3      525.0   675.0   425.0  725.0   Yes    -
  4      -15.0   -15.0   425.0  725.0   Yes    -

```

////////// Recap of all Cases //////////

(Case Number 1)OK
 (Case Number 2)OK
 (Case Number 3)OK
 (Case Number 4)OK
 (hr Calc StatTS)OK

//////////////////////////////////// Tubesheet Stresses //////////////////////////////////////

Case#	(Per.Sigma/Allow)	(Sigma)	(Per.Tau/Allow)	(Tau)
[1]	[76]	24928 < 32900	[34]	4443 < 13160
[2]	[97]	31852 < 32900	[43]	5678 < 13160
[3]	[21]	6924 < 32900	[9]	1234 < 13160

>>>> UHX-10 (b) Data for Stationary Tubesheet <<<<

Tubesheet Thicknes h = 6.2500
 Tubesheet OD = 49.6250
 A = 39.8750, at Min. Cylinder Inside Diameter turns
 Thickness of Tubesheet at A diameter = 6.2500
 Thickness at A diameter not less than Smaller of (0.75h , h-0.375)
 (4.6875 , 5.8750)
 Thickness at A diameter 6.2500 not less than 4.6875
 Tubesheet complies with UHX-10 (b) [[Yes]]

ASME Horizontal Vessel Analysis: Stresses for the Left Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Operating Case

Minimum Wear Plate Width to be considered in analysis [b1]:

= min(b + 1.56*sqrt(Rm * t), 2a)
= min(8.000 + 1.56*sqrt(20.6875 * 1.1250), 2 * 27.750)
= 15.5258 in.

Input and Calculated Values:

Vessel Mean Radius	Rm	20.69	in.
Stiffened Vessel Length per 4.15.6	L	25.29	ft.
Distance from Saddle to Vessel tangent	a	27.75	in.
Saddle Width	b	8.00	in.
Saddle Bearing Angle	theta	117.63	degrees
Shell Allowable Stress used in Calculation		16450.00	psi
Head Allowable Stress used in Calculation		0.00	psi
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Operating Case		28822.54	lb.

Horizontal Vessel Analysis Results:	Actual	Allowable	

Long. Stress at Top of Midspan	5277.87	16450.00	psi
Long. Stress at Bottom of Midspan	7138.80	16450.00	psi
Long. Stress at Top of Saddles	6547.62	16450.00	psi
Long. Stress at Bottom of Saddles	6020.78	16450.00	psi
Tangential Shear in Shell	1225.18	13160.00	psi
Circ. Stress at Horn of Saddle	2286.92	20562.50	psi
Circ. Compressive Stress in Shell	126.97	16450.00	psi

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

= Ftr * (Ft/Num of Saddles + Z Force Load) * B / E
= 3.00 * (1646.5/2 + 0) * 60.0000/41.0000
= 3614.3 lb.

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

= max(Fl, Friction Load, Sum of X Forces) * B / Ls
= max(355.22 , 0.00 , 27750) * 60.0000/141.0004
= 11808.5 lb.

Saddle Reaction Force due to Earthquake Fl or Friction [Fsl]:

= max(Fl, Friction Force, Sum of X Forces) * B / Ls
= max(2082.35 , 0.00 , 27750) * 60.0000/141.0004
= 11808.5 lb.

Saddle Reaction Force due to Earthquake Ft [Fst]:

= Ftr * (Ft/Num of Saddles + Z Force Load) * B / E
= 3.00 * (2082/2 + 0) * 60.0000/41.0000
= 4571.0 lb.

Load Combination Results for Q + Wind or Seismic [Q]:

= Saddle Load + Max(Fwl, Fwt, Fsl, Fst)
= 17014 + Max(11808 , 3614 , 11808 , 4571)
= 28822.5 lb.

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	29766.24 lb.
Transverse Shear Load Saddle	1041.18 lb.
Longitudinal Shear Load Saddle	27750.00 lb.

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, $k = 0.1$

The Computed K values from Table 4.15.1:

K1 = 0.1029	K2 = 1.2107	K3 = 0.9225	K4 = 0.4110
K5 = 0.7694	K6 = 0.0549	K7 = 0.0549	K8 = 0.3427
K9 = 0.2746	K10 = 0.0598	K1* = 0.1862	

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned}
 &= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right] \\
 &= -28823 \cdot 2.31 \left[1 - \left(1 - \frac{2.31}{25.29} + \frac{(1.724^2 - 0.000^2)}{(2 \cdot 2.31 \cdot 25.29)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 25.29)} \right) \right] \\
 &= -4400.7 \text{ ft}\cdot\text{lb.}
 \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned}
 &= Q \cdot L / 4 \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{(3L)} \right) - 4a / L \\
 &= 28823 \cdot 25.3 / 4 \left(1 + 2 \frac{(1.724^2 - 0.000^2)}{(25.29^2)} \right) / \left(1 + \frac{(4 \cdot 0.000)}{(3 \cdot 25.29)} \right) - 4 \cdot 2.31 / 25.29 \\
 &= 117283.9 \text{ ft}\cdot\text{lb.}
 \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M_2 / (\pi \cdot R_m^2 t) \\
 &= 675.227 \cdot 20.688 / (2 \cdot 1.125) - 1407407 / (\pi \cdot 20.7^2 \cdot 1.125) \\
 &= 5277.87 \text{ psi}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M_2 / (\pi \cdot R_m^2 \cdot t) \\
 &= 675.227 \cdot 20.688 / (2 \cdot 1.125) + 1407407 / (\pi \cdot 20.7^2 \cdot 1.125) \\
 &= 7138.80 \text{ psi}
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M_1 / (K_1 \cdot \pi \cdot R_m^2 t) \\
 &= 675.227 \cdot 20.688 / (2 \cdot 1.125) - 52809.0 / (0.1029 \cdot \pi \cdot 20.7^2 \cdot 1.125) \\
 &= 6547.62 \text{ psi}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M_1 / (K_1 \cdot \pi \cdot R_m^2 \cdot t) \\
 &= 675.227 \cdot 20.688 / (2 \cdot 1.125) + 52809.0 / (0.1862 \cdot \pi \cdot 20.7^2 \cdot 1.125) \\
 &= 6020.78 \text{ psi}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L - 2a) / (L + (4 \cdot h^2 / 3)) \\
 &= 28823 (25.29 - 2 \cdot 2.31) / (25.29 + (4 \cdot 0.00 / 3)) \\
 &= 23551.9 \text{ lb.}
 \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned}
 &= K_2 \cdot T / (R_m \cdot t) \\
 &= 1.2107 \cdot 23551.86 / (20.6875 \cdot 1.1250) \\
 &= 1225.18 \text{ psi}
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 \cdot \sqrt{R_m \cdot t} \\
 &= 0.78 \cdot \sqrt{20.688 \cdot 1.125} \\
 &= 3.763 \text{ in.}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$= -K5 * Q * k / (t * (b + X1 + X2))$$

$$= -0.7694 * 28823 * 0.1 / (1.125 * (8.00 + 3.76 + 3.76))$$

$$= -126.97 \text{ psi}$$

Circ. Comp. Stress at Horn of Saddle, L>=8Rm (4.15.24) [sigma7]:

$$= -Q / (4 * t * (b + X1 + X2)) - 3 * K7 * Q / (2 * t^2)$$

$$= -28823 / (4 * 1.125 * (8.000 + 3.763 + 3.763)) -$$

$$3 * 0.0549 * 28823 / (2 * 1.125^2)$$

$$= -2286.92 \text{ psi}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a)$$

$$= \min(8.00 + 1.56 * \text{sqrt}(20.688 * 1.125), 2 * 27.750)$$

$$= 15.53 \text{ in.}$$

Free Un-Restrained Thermal Expansion between the Saddles [Exp]:

$$= \text{Alpha} * Ls * (\text{Design Temperature} - \text{Ambient Temperature})$$

$$= 0.765E-05 * 141.000 * (725.0 - 70.0)$$

$$= 0.707 \text{ in.}$$

Results for Vessel Ribs, Web and Base:

Baseplate Length	Bplen	41.0000	in.
Baseplate Thickness	Bpthk	0.6250	in.
Baseplate Width	Bpwid	9.0000	in.
Number of Ribs (inc. outside ribs)	Nribs	2	
Rib Thickness	Ribtk	1.2500	in.
Web Thickness	Webtk	1.1250	in.
Web Location	Webloc	Center	

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	0.5625	19.6007	11.0254	8.2690
Wearplate	1.3125	3.7500	4.9219	6.5039
Web	20.4375	42.6094	870.8291	22891.2227
BasePlate	39.6875	5.6250	223.2422	8860.1074
Totals	62.0000	71.5850	1110.0186	31766.1035

Value C1 = Sumof(Ay) / Sumof(A) = 15.5063 in.

Value I = Sumof(Io) - C1 * Sumof(Ay) = 14553.8320 in**4

Value As = Sumof(A) - Ashell = 51.9844 in²

$$K1 = (1 + \text{Cos}(\text{beta}) - .5 * \text{Sin}(\text{beta})^2) / (\text{pi} - \text{beta} + \text{Sin}(\text{beta}) * \text{Cos}(\text{beta})) = 0.1992$$

$$Fh = K1 * Q = 0.1992 * 28822.543 = 5742.1738 \text{ lb.}$$

Tension Stress, St = (Fh / As) = 110.4596 psi

Allowed Stress, Sa = 0.6 * Yield Str = 21600.0000 psi

d = B - R * Sin(theta) / theta = 43.3527 in.

Bending Moment, M = Fh * d = 20744.8848 ft.lb.

Bending Stress, Sb = (M * C1 / I) = 265.2301 psi

Allowed Stress, Sa = 2/3 * Yield Str = 24000.0000 psi

Minimum Thickness of Baseplate per Moss :

$$= (3 * (Q + \text{Saddle_Wt}) * \text{BasePlateWidth} / (4 * \text{BasePlateLength} *$$

$$\text{AllStress}))^{1/2}$$

$$= (3 * (28823 + 944) * 9.00 / (4 * 41.000 * 24000.000))^{1/2}$$

$$= 0.452 \text{ in.}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress

Effective Baseplate Length [e]:

$$= (B_{plen} - Clearance) / (N_{ribs} - 1) \\ = (41.0000 - 1.0) / (2 - 1) = 40.0000 \text{ in.}$$

Baseplate Pressure Area [A_p]:

$$= e * B_{pwid} / 2 \\ = 40.0000 * 9.0000 / 2 = 180.0000 \text{ in}^2$$

Axial Load [P]:

$$= A_p * B_p \\ = 180.0 * 78.11 = 14059.8 \text{ lb.}$$

Area of the Rib and Web [A_r]:

$$= (B_{pwid} - Clearance - Web_{tk}) * Rib_{tk} + e/2 * Web_{tk} \\ = (9.000 - 1.0 - 1.125) * 1.250 + 40.0000/2 * 1.125 \\ = 31.094 \text{ in}^2$$

Compressive Stress [S_c]:

$$= P/A_r \\ = 14059.8/31.0938 = 452.1737 \text{ psi}$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay ²	Io
Rib	4.0000	8.5938	34.3750	0.0000	53.3333
Web	4.0000	22.5000	90.0000	0.0000	4.7461
Values	4.0000	31.0938	124.3750	0.0000	58.0794

Bending Moment [R_m]:

$$= F_l / (2 * B_{plen}) * e * r_l / 2 \\ = 27750.0 / (2 * 41.00) * 40.000 * 49.12/2 \\ = 27705.650 \text{ ft.lb.}$$

KL/R < C_c (35.9415 < 126.0993) per AISC E2-1

$$S_{ca} = (1 - (K_{lr})^2 / (2 * C_c^2)) * F_y / (5/3 + 3 * (K_{lr}) / (8 * C_c) - (K_{lr}^3) / (8 * C_c^3))$$

$$S_{ca} = (1 - (35.94)^2 / (2 * 126.10^2)) * 36000 / \\ (5/3 + 3 * (35.94) / (8 * 126.10) - (35.94^3) / (8 * 126.10^3))$$

$$S_{ca} = 19505.58 \text{ psi}$$

AISC Unity Check on Outside Ribs (must be <= 1.0)

$$Check = S_c/S_{ca} + (R_m/Z) / S_{ba}$$

$$Check = 452.17/19505.58 + (332467.81/14.520) / 24000.00$$

$$Check = 0.98$$

ASME Horizontal Vessel Analysis: Stresses for the Right Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Minimum Wear Plate Width to be considered in analysis [b₁]:

$$= \min(b + 1.56 * \sqrt{ R_m * t }, 2a) \\ = \min(8.000 + 1.56 * \sqrt{ 20.6875 * 1.1250 }, 2 * 69.000) \\ = 15.5258 \text{ in.}$$

Input and Calculated Values:

Vessel Mean Radius	Rm	20.69	in.
Stiffened Vessel Length per 4.15.6	L	25.29	ft.
Distance from Saddle to Vessel tangent	a	69.00	in.
Saddle Width	b	8.00	in.
Saddle Bearing Angle	theta	117.63	degrees
Shell Allowable Stress used in Calculation		16450.00	psi
Head Allowable Stress used in Calculation		0.00	psi
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Operating Case		37473.15	lb.

Horizontal Vessel Analysis Results:	Actual	Allowable	

Long. Stress at Top of Midspan	6020.54	16450.00	psi
Long. Stress at Bottom of Midspan	6396.12	16450.00	psi
Long. Stress at Top of Saddles	9815.31	16450.00	psi
Long. Stress at Bottom of Saddles	4214.45	16450.00	psi
Tangential Shear in Shell	1063.00	13160.00	psi
Circ. Stress at Horn of Saddle	2973.30	20562.50	psi
Circ. Compressive Stress in Shell	165.07	16450.00	psi

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$\begin{aligned}
 &= F_{tr} * (Ft/Num\ of\ Saddles + Z\ Force\ Load) * B / E \\
 &= 3.00 * (1646.5/2 + 0) * 60.0000/41.0000 \\
 &= 3614.3\ lb.
 \end{aligned}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$\begin{aligned}
 &= \max(F_l, Friction\ Load, Sum\ of\ X\ Forces) * B / L_s \\
 &= \max(355.22 , 0.00 , 27750) * 60.0000/141.0004 \\
 &= 11808.5\ lb.
 \end{aligned}$$

Saddle Reaction Force due to Earthquake Fl or Friction [Fsl]:

$$\begin{aligned}
 &= \max(F_l, Friction\ Force, Sum\ of\ X\ Forces) * B / L_s \\
 &= \max(2082.35 , 0.00 , 27750) * 60.0000/141.0004 \\
 &= 11808.5\ lb.
 \end{aligned}$$

Saddle Reaction Force due to Earthquake Ft [Fst]:

$$\begin{aligned}
 &= F_{tr} * (Ft/Num\ of\ Saddles + Z\ Force\ Load) * B / E \\
 &= 3.00 * (2082/2 + 0) * 60.0000/41.0000 \\
 &= 4571.0\ lb.
 \end{aligned}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$\begin{aligned}
 &= Saddle\ Load + \max(F_{wl}, F_{wt}, F_{sl}, F_{st}) \\
 &= 25665 + \max(11808 , 3614 , 11808 , 4571) \\
 &= 37473.1\ lb.
 \end{aligned}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	38416.84	lb.
Transverse Shear Load Saddle	1041.18	lb.
Longitudinal Shear Load Saddle	27750.00	lb.

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, $k = 0.1$

The Computed K values from Table 4.15.1:

$$\begin{aligned} K1 &= 0.1029 & K2 &= 1.2107 & K3 &= 0.9225 & K4 &= 0.4110 \\ K5 &= 0.7694 & K6 &= 0.0549 & K7 &= 0.0549 & K8 &= 0.3427 \\ K9 &= 0.2746 & K10 &= 0.0598 & K1^* &= 0.1862 & & \end{aligned}$$

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned} &= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{(3L)} \right) \right] \\ &= -37473 \cdot 5.75 \left[1 - \left(1 - \frac{5.75}{25.29} + \frac{(1.724^2 - 0.000^2)}{(2 \cdot 5.75 \cdot 25.29)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 25.29)} \right) \right] \\ &= -46785.0 \text{ ft.lb.} \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned} &= Q \cdot L / 4 \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{(3L)} \right) - 4a / L \\ &= 37473 \cdot 25.3 / 4 \left(1 + 2 \frac{(1.724^2 - 0.000^2)}{(25.29^2)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 25.29)} \right) - 4 \cdot 5.75 / 25.29 \\ &= 23670.7 \text{ ft.lb.} \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned} &= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 \cdot t) \\ &= 675.227 \cdot 20.688 / (2 \cdot 1.125) - 284048.8 / (\pi \cdot 20.7^2 \cdot 1.125) \\ &= 6020.54 \text{ psi} \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned} &= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t) \\ &= 675.227 \cdot 20.688 / (2 \cdot 1.125) + 284048.8 / (\pi \cdot 20.7^2 \cdot 1.125) \\ &= 6396.12 \text{ psi} \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned} &= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\ &= 675.227 \cdot 20.688 / (2 \cdot 1.125) - 561419.9 / (0.1029 \cdot \pi \cdot 20.7^2 \cdot 1.125) \\ &= 9815.31 \text{ psi} \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned} &= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\ &= 675.227 \cdot 20.688 / (2 \cdot 1.125) + 561419.9 / (0.1862 \cdot \pi \cdot 20.7^2 \cdot 1.125) \\ &= 4214.45 \text{ psi} \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned} &= Q (L - 2a) / \left(L + \frac{(4h^2)}{3} \right) \\ &= 37473 \left(25.29 - 2 \cdot 5.75 \right) / \left(25.29 + \frac{(4 \cdot 0.00)}{3} \right) \\ &= 20434.3 \text{ lb.} \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned} &= K2 \cdot T / (R_m \cdot t) \\ &= 1.2107 \cdot 20434.29 / (20.6875 \cdot 1.125) \\ &= 1063.00 \text{ psi} \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned} &= 0.78 \cdot \sqrt{R_m \cdot t} \\ &= 0.78 \cdot \sqrt{20.688 \cdot 1.125} \\ &= 3.763 \text{ in.} \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned} &= -K5 \cdot Q \cdot k / (t \cdot (b + X1 + X2)) \\ &= -0.7694 \cdot 37473 \cdot 0.1 / (1.125 \cdot (8.00 + 3.76 + 3.76)) \\ &= -165.07 \text{ psi} \end{aligned}$$

Circ. Comp. Stress at Horn of Saddle, $L \geq 8R_m$ (4.15.24) [σ_7]:

$$= -Q / (4 * t * (b + X_1 + X_2)) - 3 * K_7 * Q / (2 * t^2)$$

$$= -37473 / (4 * 1.125 * (8.000 + 3.763 + 3.763)) - 3 * 0.0549 * 37473 / (2 * 1.125^2)$$

$$= -2973.30 \text{ psi}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$= \min(b + 1.56 * \sqrt{ R_m * t }, 2a)$$

$$= \min(8.00 + 1.56 * \sqrt{ 20.688 * 1.125 }, 2 * 69.000)$$

$$= 15.53 \text{ in.}$$

Results for Vessel Ribs, Web and Base

Baseplate Length	Bplen	41.0000	in.
Baseplate Thickness	Bpthk	0.6250	in.
Baseplate Width	Bpwid	9.0000	in.
Number of Ribs (inc. outside ribs)	Nribs	2	
Rib Thickness	Ribtk	1.2500	in.
Web Thickness	Webtk	1.1250	in.
Web Location	Webloc	Center	

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	0.5625	19.6007	11.0254	8.2690
Wearplate	1.3125	3.7500	4.9219	6.5039
Web	20.4375	42.6094	870.8291	22891.2227
BasePlate	39.6875	5.6250	223.2422	8860.1074
Totals	62.0000	71.5850	1110.0186	31766.1035

$$\text{Value } C_1 = \text{Sumof}(AY) / \text{Sumof}(A) = 15.5063 \text{ in.}$$

$$\text{Value } I = \text{Sumof}(Io) - C_1 * \text{Sumof}(AY) = 14553.8320 \text{ in}^4$$

$$\text{Value } A_s = \text{Sumof}(A) - A_{\text{shell}} = 51.9844 \text{ in}^2$$

$$K_1 = (1 + \cos(\beta) - 0.5 \sin(\beta)^2) / (\pi - \beta + \sin(\beta) * \cos(\beta)) = 0.1992$$

$$F_h = K_1 * Q = 0.1992 * 37473.148 = 7465.5918 \text{ lb.}$$

$$\text{Tension Stress, } S_t = (F_h / A_s) = 143.6122 \text{ psi}$$

$$\text{Allowed Stress, } S_a = 0.6 * \text{Yield Str} = 21600.0000 \text{ psi}$$

$$d = B - R * \sin(\theta) / \theta = 43.3527 \text{ in.}$$

$$\text{Bending Moment, } M = F_h * d = 26971.1172 \text{ ft.lb.}$$

$$\text{Bending Stress, } S_b = (M * C_1 / I) = 344.8345 \text{ psi}$$

$$\text{Allowed Stress, } S_a = 2/3 * \text{Yield Str} = 24000.0000 \text{ psi}$$

Minimum Thickness of Baseplate per Moss :

$$= (3 * (Q + \text{Saddle_Wt}) * \text{BasePlateWidth} / (4 * \text{BasePlateLength} * \text{AllStress}))^{1/2}$$

$$= (3 * (37473 + 944) * 9.00 / (4 * 41.000 * 24000.000))^{1/2}$$

$$= 0.513 \text{ in.}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress

Effective Baseplate Length [e]:

$$= (Bplen - \text{Clearance}) / (Nribs - 1)$$

$$= (41.0000 - 1.0) / (2 - 1) = 40.0000 \text{ in.}$$

Baseplate Pressure Area [Ap]:

$$= e * Bpwid / 2$$

$$= 40.0000 * 9.0000 / 2 = 180.0000 \text{ in}^2$$

Axial Load [P]:

$$= A_p * B_p$$

$$= 180.0 * 101.55 = 18279.6 \text{ lb.}$$

Area of the Rib and Web [Ar]:

$$= (B_{pwid} - Clearance - Webtk) * Ribtk + e/2 * Webtk$$

$$= (9.000 - 1.0 - 1.125) * 1.250 + 40.0000/2 * 1.125$$

$$= 31.094 \text{ in}^2$$

Compressive Stress [Sc]:

$$= P/Ar$$

$$= 18279.6/31.0938 = 587.8861 \text{ psi}$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay ²	Io
Rib	4.0000	8.5938	34.3750	0.0000	53.3333
Web	4.0000	22.5000	90.0000	0.0000	4.7461
Values	4.0000	31.0938	124.3750	0.0000	58.0794

Bending Moment [Rm]:

$$= F_l / (2 * B_{plen}) * e * r_l / 2$$

$$= 27750.0 / (2 * 41.00) * 40.000 * 49.12/2$$

$$= 27705.650 \text{ ft.lb.}$$

KL/R < Cc (35.9415 < 126.0993) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = (1 - (35.94)^2 / (2 * 126.10^2)) * 36000 /$$

$$(5/3 + 3 * (35.94) / (8 * 126.10) - (35.94^3) / (8 * 126.10^3))$$

$$Sca = 19505.58 \text{ psi}$$

AISC Unity Check on Outside Ribs (must be <= 1.0)

$$Check = Sc/Sca + (Rm/Z) / Sba$$

$$Check = 587.89/19505.58 + (332467.81/14.520) / 24000.00$$

$$Check = 0.98$$

PV Elite is a trademark of Intergraph CADWorx & Analysis Solutions, Inc. 2015

ASME Horizontal Vessel Analysis: Stresses for the Left Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Test Case

Minimum Wear Plate Width to be considered in analysis [b1]:

$$= \min(b + 1.56 \cdot \sqrt{ R_m \cdot t }, 2a)$$

$$= \min(8.000 + 1.56 \cdot \sqrt{ 20.6250 \cdot 1.2500 }, 2 \cdot 27.750)$$

$$= 15.9209 \text{ in.}$$

Input and Calculated Values:

Vessel Mean Radius	Rm	20.62	in.
Stiffened Vessel Length per 4.15.6	L	25.29	ft.
Distance from Saddle to Vessel tangent	a	27.75	in.
Saddle Width	b	8.00	in.
Saddle Bearing Angle	theta	117.63	degrees
Shell Allowable Stress used in Calculation		36100.00	psi
Head Allowable Stress used in Calculation		36100.00	psi
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Test Case, no Ext. Forces		26513.40	lb.

Horizontal Vessel Analysis Results:	Actual	Allowable	

Long. Stress at Top of Midspan	6466.46	36100.00	psi
Long. Stress at Bottom of Midspan	8016.33	36100.00	psi
Long. Stress at Top of Saddles	7524.65	36100.00	psi
Long. Stress at Bottom of Saddles	7084.82	36100.00	psi
Tangential Shear in Shell	1017.39	28880.00	psi
Circ. Stress at Horn of Saddle	1729.68	54150.00	psi
Circ. Compressive Stress in Shell	102.51	36100.00	psi

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= F_{tr} \cdot (Ft / \text{Num of Saddles} + Z \text{ Force Load}) \cdot B / E$$

$$= 3.00 \cdot (543.4 / 2 + 0) \cdot 60.0000 / 41.0000$$

$$= 1192.7 \text{ lb.}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= \max(F_l, \text{Friction Load, Sum of X Forces}) \cdot B / L_s$$

$$= \max(117.22, 0.00, 0) \cdot 60.0000 / 141.0004$$

$$= 49.9 \text{ lb.}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$= \text{Saddle Load} + \max(F_{wl}, F_{wt}, F_{sl}, F_{st})$$

$$= 25321 + \max(50, 1193, 0, 0)$$

$$= 26513.4 \text{ lb.}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	27457.10	lb.
Transverse Shear Load Saddle	271.68	lb.
Longitudinal Shear Load Saddle	117.22	lb.

Hydrostatic Test Pressure at center of Vessel: 877.745 psig

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, $k = 0.1$

The Computed K values from Table 4.15.1:

$$\begin{aligned} K1 &= 0.1029 & K2 &= 1.2107 & K3 &= 0.9225 & K4 &= 0.4110 \\ K5 &= 0.7694 & K6 &= 0.0549 & K7 &= 0.0549 & K8 &= 0.3427 \\ K9 &= 0.2746 & K10 &= 0.0598 & K1^* &= 0.1862 \end{aligned}$$

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned} &= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{(3L)} \right) \right] \\ &= -26513 \cdot 2.31 \left[1 - \left(1 - \frac{2.31}{25.29} + \frac{(1.719^2 - 0.000^2)}{(2 \cdot 2.31 \cdot 25.29)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 25.29)} \right) \right] \\ &= -4057.6 \text{ ft.lb.} \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned} &= Q \cdot L / 4 \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{(3L)} \right) - 4a/L \\ &= 26513 \cdot 25.3 / 4 \left(1 + 2 \frac{(1.719^2 - 0.000^2)}{(25.29^2)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 25.29)} \right) - 4 \cdot 2.31 / 25.29 \\ &= 107878.2 \text{ ft.lb.} \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned} &= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 \cdot t) \\ &= 877.745 \cdot 20.625 / (2 \cdot 1.250) - 1294539 / (\pi \cdot 20.6^2 \cdot 1.250) \\ &= 6466.46 \text{ psi} \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned} &= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t) \\ &= 877.745 \cdot 20.625 / (2 \cdot 1.250) + 1294539 / (\pi \cdot 20.6^2 \cdot 1.250) \\ &= 8016.33 \text{ psi} \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned} &= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\ &= 877.745 \cdot 20.625 / (2 \cdot 1.250) - 48690.9 / (0.1029 \cdot \pi \cdot 20.6^2 \cdot 1.250) \\ &= 7524.65 \text{ psi} \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned} &= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\ &= 877.745 \cdot 20.625 / (2 \cdot 1.250) + 48690.9 / (0.1862 \cdot \pi \cdot 20.6^2 \cdot 1.250) \\ &= 7084.82 \text{ psi} \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned} &= Q(L - 2a) / (L + (4 \cdot h^2 / 3)) \\ &= 26513 (25.29 - 2 \cdot 2.31) / (25.29 + (4 \cdot 0.00 / 3)) \\ &= 21665.0 \text{ lb.} \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned} &= K2 \cdot T / (R_m \cdot t) \\ &= 1.2107 \cdot 21664.99 / (20.6250 \cdot 1.2500) \\ &= 1017.39 \text{ psi} \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned} &= 0.78 \cdot \sqrt{R_m \cdot t} \\ &= 0.78 \cdot \sqrt{20.625 \cdot 1.250} \\ &= 3.960 \text{ in.} \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned} &= -K5 \cdot Q \cdot k / (t \cdot (b + X1 + X2)) \\ &= -0.7694 \cdot 26513 \cdot 0.1 / (1.250 \cdot (8.00 + 3.96 + 3.96)) \\ &= -102.51 \text{ psi} \end{aligned}$$

Circ. Comp. Stress at Horn of Saddle, L>=8Rm (4.15.24) [sigma7]:

$$\begin{aligned}
&= -Q/(4*t*(b+X1+X2)) - 3*K7*Q/(2*t^2) \\
&= -26513/(4*1.250*(8.000+3.960+3.960)) - \\
&\quad 3*0.0549*26513/(2*1.250^2) \\
&= -1729.68 \text{ psi}
\end{aligned}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$\begin{aligned}
&= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a) \\
&= \min(8.00 + 1.56 * \text{sqrt}(20.625 * 1.250), 2 * 27.750) \\
&= 15.92 \text{ in.}
\end{aligned}$$

Results for Vessel Ribs, Web and Base:

Baseplate Length	Bplen	41.0000	in.
Baseplate Thickness	Bpthk	0.6250	in.
Baseplate Width	Bpwid	9.0000	in.
Number of Ribs (inc. outside ribs)	Nribs	2	
Rib Thickness	Ribtk	1.2500	in.
Web Thickness	Webtk	1.1250	in.
Web Location	Webloc	Center	

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	0.6250	22.2500	13.9062	11.5885
Wearplate	1.4375	3.7500	5.3906	7.7930
Web	20.5000	42.4688	870.6094	22890.8770
BasePlate	39.6875	5.6250	223.2422	8860.1074
Totals	62.2500	74.0938	1113.1484	31770.3652

$$\begin{aligned}
\text{Value } C1 &= \text{Sumof}(Ay) / \text{Sumof}(A) &= & 15.0235 \text{ in.} \\
\text{Value } I &= \text{Sumof}(Io) - C1 * \text{Sumof}(Ay) &= & 15046.9668 \text{ in}^{*4} \\
\text{Value } As &= \text{Sumof}(A) - A_{shell} &= & 51.8438 \text{ in}^2
\end{aligned}$$

$$K1 = (1 + \cos(\beta) - .5 * \sin(\beta)^2) / (\pi - \beta + \sin(\beta) * \cos(\beta)) = 0.1992$$

$$Fh = K1 * Q = 0.1992 * 26513.402 = 5282.1353 \text{ lb.}$$

$$\begin{aligned}
\text{Tension Stress, } St &= (Fh / As) &= & 101.8857 \text{ psi} \\
\text{Allowed Stress, } Sa &= 0.6 * \text{Yield Str} &= & 21600.0000 \text{ psi}
\end{aligned}$$

$$\begin{aligned}
d &= B - R * \sin(\theta) / \theta &= & 43.3319 \text{ in.} \\
\text{Bending Moment, } M &= Fh * d &= & 19073.7266 \text{ ft.lb.}
\end{aligned}$$

$$\begin{aligned}
\text{Bending Stress, } Sb &= (M * C1 / I) &= & 228.5280 \text{ psi} \\
\text{Allowed Stress, } Sa &= 2/3 * \text{Yield Str} &= & 24000.0000 \text{ psi}
\end{aligned}$$

Minimum Thickness of Baseplate per Moss :

$$\begin{aligned}
&= (3 * (Q + \text{Saddle_Wt}) * \text{BasePlateWidth} / (4 * \text{BasePlateLength} * \\
&\quad \text{AllStress}))^{1/2} \\
&= (3 * (26513 + 944) * 9.00 / (4 * 41.000 * 24000.000))^{1/2} \\
&= 0.434 \text{ in.}
\end{aligned}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress

Effective Baseplate Length [e]:

$$\begin{aligned}
&= (Bplen - \text{Clearance}) / (Nribs - 1) \\
&= (41.0000 - 1.0) / (2 - 1) = 40.0000 \text{ in.}
\end{aligned}$$

Baseplate Pressure Area [Ap]:

$$\begin{aligned}
&= e * Bpwid / 2 \\
&= 40.0000 * 9.0000 / 2 = 180.0000 \text{ in}^2
\end{aligned}$$

Axial Load [P]:

$$= A_p * B_p$$

$$= 180.0 * 71.85 = 12933.4 \text{ lb.}$$

Area of the Rib and Web [Ar]:

$$= (B_{pwid} - Clearance - Webtk) * Ribtk + e/2 * Webtk$$

$$= (9.000 - 1.0 - 1.125) * 1.250 + 40.0000/2 * 1.125$$

$$= 31.094 \text{ in}^2$$

Compressive Stress [Sc]:

$$= P/Ar$$

$$= 12933.4/31.0938 = 415.9474 \text{ psi}$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay ²	Io
Rib	4.0000	8.5938	34.3750	0.0000	53.3333
Web	4.0000	22.5000	90.0000	0.0000	4.7461
Values	4.0000	31.0938	124.3750	0.0000	58.0794

Bending Moment [Rm]:

$$= F_l / (2 * B_{plen}) * e * r_l / 2$$

$$= 117.2 / (2 * 41.00) * 40.000 * 49.00/2$$

$$= 116.738 \text{ ft.lb.}$$

KL/R < Cc (35.8500 < 126.0993) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = (1 - (35.85)^2 / (2 * 126.10^2)) * 36000 / (5/3 + 3 * (35.85) / (8 * 126.10) - (35.85^3) / (8 * 126.10^3))$$

$$Sca = 19512.54 \text{ psi}$$

AISC Unity Check on Outside Ribs (must be <= 1.0)

$$Check = Sc/Sca + (Rm/Z)/Sba$$

$$Check = 415.95/19512.54 + (1400.86/14.520)/24000.00$$

$$Check = 0.03$$

ASME Horizontal Vessel Analysis: Stresses for the Right Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Minimum Wear Plate Width to be considered in analysis [b1]:

$$= \min(b + 1.56 * \sqrt{ Rm * t }, 2a)$$

$$= \min(8.000 + 1.56 * \sqrt{ 20.6250 * 1.2500 }, 2 * 69.000)$$

$$= 15.9209 \text{ in.}$$

Input and Calculated Values:

Vessel Mean Radius	Rm	20.62	in.
Stiffened Vessel Length per 4.15.6	L	25.29	ft.
Distance from Saddle to Vessel tangent	a	69.00	in.
Saddle Width	b	8.00	in.
Saddle Bearing Angle	theta	117.63	degrees
Shell Allowable Stress used in Calculation		36100.00	psi
Head Allowable Stress used in Calculation		36100.00	psi
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Test Case, no Ext. Forces		27890.93	lb.

Horizontal Vessel Analysis Results:	Actual	Allowable	
Long. Stress at Top of Midspan	7114.91	36100.00	psi
Long. Stress at Bottom of Midspan	7367.88	36100.00	psi
Long. Stress at Top of Saddles	9672.93	36100.00	psi
Long. Stress at Bottom of Saddles	5897.28	36100.00	psi
Tangential Shear in Shell	714.22	28880.00	psi
Circ. Stress at Horn of Saddle	1819.54	54150.00	psi
Circ. Compressive Stress in Shell	107.83	36100.00	psi

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= F_{tr} * (Ft/Num\ of\ Saddles + Z\ Force\ Load) * B / E$$

$$= 3.00 * (543.4/2 + 0) * 60.0000/41.0000$$

$$= 1192.7\ lb.$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= \max(F_l, Friction\ Load, Sum\ of\ X\ Forces) * B / L_s$$

$$= \max(117.22 , 0.00 , 0) * 60.0000/141.0004$$

$$= 49.9\ lb.$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$= Saddle\ Load + \max(F_{wl}, F_{wt}, F_{sl}, F_{st})$$

$$= 26698 + \max(50 , 1193 , 0 , 0)$$

$$= 27890.9\ lb.$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	28834.63	lb.
Transverse Shear Load Saddle	271.68	lb.
Longitudinal Shear Load Saddle	117.22	lb.

Hydrostatic Test Pressure at center of Vessel: 877.745 psig

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, k = 0.1

The Computed K values from Table 4.15.1:

K1 = 0.1029	K2 = 1.2107	K3 = 0.9225	K4 = 0.4110
K5 = 0.7694	K6 = 0.0549	K7 = 0.0549	K8 = 0.3427
K9 = 0.2746	K10 = 0.0598	K1* = 0.1862	

Note: Dimension a is greater than or equal to Rm / 2.

Moment per Equation 4.15.3 [M1]:

$$= -Q*a [1 - (1 - a/L + (R^2-h^2)/(2a*L)) / (1 + (4h^2)/3L)]$$

$$= -27891*5.75 [1 - (1 - 5.75/25.29 + (1.719^2 - 0.000^2) / (2*5.75*25.29)) / (1 + (4*0.00) / (3*25.29))]$$

$$= -34831.5\ ft.lb.$$

Moment per Equation 4.15.4 [M2]:

$$= Q*L/4 (1 + 2(R^2-h^2)/(L^2)) / (1 + (4h^2)/(3L)) - 4a/L$$

$$= 27891*25.3/4 (1 + 2(1.719^2 - 0.000^2)/(25.29^2)) / (1 + (4*0.000)/(3*25.292)) - 4*5.75/25.29$$

$$= 17608.0\ ft.lb.$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$= P * R_m / (2t) - M_2 / (\pi * R_m^2 t)$$

$$= 877.745 * 20.625 / (2*1.250) - 211296.5 / (\pi * 20.6^2 * 1.250)$$

$$= 7114.91\ psi$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
&= P * Rm / (2t) + M2 / (\pi * Rm^2 * t) \\
&= 877.745 * 20.625 / (2 * 1.250) + 211296.5 / (\pi * 20.6^2 * 1.250) \\
&= 7367.88 \text{ psi}
\end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned}
&= P * Rm / (2t) - M1 / (K1 * \pi * Rm^2 * t) \\
&= 877.745 * 20.625 / (2 * 1.250) - 417978.5 / (0.1029 * \pi * 20.6^2 * 1.250) \\
&= 9672.93 \text{ psi}
\end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned}
&= P * Rm / (2t) + M1 / (K1 * \pi * Rm^2 * t) \\
&= 877.745 * 20.625 / (2 * 1.250) + 417978.5 / (0.1862 * \pi * 20.6^2 * 1.250) \\
&= 5897.28 \text{ psi}
\end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
&= Q(L-2a) / (L + (4 * h^2 / 3)) \\
&= 27891 (25.29 - 2 * 5.75) / (25.29 + (4 * 0.00 / 3)) \\
&= 15209.1 \text{ lb.}
\end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned}
&= K2 * T / (Rm * t) \\
&= 1.2107 * 15209.06 / (20.6250 * 1.2500) \\
&= 714.22 \text{ psi}
\end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
&= 0.78 * \text{sqrt}(Rm * t) \\
&= 0.78 * \text{sqrt}(20.625 * 1.250) \\
&= 3.960 \text{ in.}
\end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
&= -K5 * Q * k / (t * (b + X1 + X2)) \\
&= -0.7694 * 27891 * 0.1 / (1.250 * (8.00 + 3.96 + 3.96)) \\
&= -107.83 \text{ psi}
\end{aligned}$$

Circ. Comp. Stress at Horn of Saddle, L>=8Rm (4.15.24) [sigma7]:

$$\begin{aligned}
&= -Q / (4 * t * (b + X1 + X2)) - 3 * K7 * Q / (2 * t^2) \\
&= -27891 / (4 * 1.250 * (8.000 + 3.960 + 3.960)) - \\
&\quad 3 * 0.0549 * 27891 / (2 * 1.250^2) \\
&= -1819.54 \text{ psi}
\end{aligned}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$\begin{aligned}
&= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a) \\
&= \min(8.00 + 1.56 * \text{sqrt}(20.625 * 1.250), 2 * 69.000) \\
&= 15.92 \text{ in.}
\end{aligned}$$

Results for Vessel Ribs, Web and Base

Baseplate Length	Bplen	41.0000	in.
Baseplate Thickness	Bpthk	0.6250	in.
Baseplate Width	Bpwid	9.0000	in.
Number of Ribs (inc. outside ribs)	Nribs	2	
Rib Thickness	Ribtk	1.2500	in.
Web Thickness	Webtk	1.1250	in.
Web Location	Webloc	Center	

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	0.6250	22.2500	13.9062	11.5885
Wearplate	1.4375	3.7500	5.3906	7.7930
Web	20.5000	42.4688	870.6094	22890.8770
BasePlate	39.6875	5.6250	223.2422	8860.1074
Totals	62.2500	74.0938	1113.1484	31770.3652

Value C1 = Sumof(Ay)/Sumof(A) = 15.0235 in.
 Value I = Sumof(Io) - C1*Sumof(Ay) = 15046.9668 in**4
 Value As = Sumof(A) - Ashell = 51.8438 in²

$K1 = (1 + \cos(\beta) - 0.5 \sin(\beta)^2) / (\pi - \beta + \sin(\beta) \cos(\beta)) = 0.1992$

$Fh = K1 * Q = 0.1992 * 27890.934 = 5556.5742 \text{ lb.}$

Tension Stress, St = (Fh/As) = 107.1793 psi
 Allowed Stress, Sa = 0.6 * Yield Str = 21600.0000 psi

$d = B - R \sin(\theta) / \theta = 43.3319 \text{ in.}$
 Bending Moment, M = Fh * d = 20064.7227 ft.lb.

Bending Stress, Sb = (M * C1 / I) = 240.4014 psi
 Allowed Stress, Sa = 2/3 * Yield Str = 24000.0000 psi

Minimum Thickness of Baseplate per Moss :

$= (3 * (Q + \text{Saddle_Wt}) * \text{BasePlateWidth} / (4 * \text{BasePlateLength} * \text{AllStress}))^{1/2}$
 $= (3 * (27891 + 944) * 9.00 / (4 * 41.000 * 24000.000))^{1/2}$
 $= 0.445 \text{ in.}$

Calculation of Axial Load, Intermediate Values and Compressive Stress

Effective Baseplate Length [e]:

$= (B_{plen} - \text{Clearance}) / (N_{ribs} - 1)$
 $= (41.0000 - 1.0) / (2 - 1) = 40.0000 \text{ in.}$

Baseplate Pressure Area [Ap]:

$= e * B_{pwid} / 2$
 $= 40.0000 * 9.0000 / 2 = 180.0000 \text{ in}^2$

Axial Load [P]:

$= A_p * B_p$
 $= 180.0 * 75.59 = 13605.3 \text{ lb.}$

Area of the Rib and Web [Ar]:

$= (B_{pwid} - \text{Clearance} - \text{Webtk}) * \text{Ribtk} + e/2 * \text{Webtk}$
 $= (9.000 - 1.0 - 1.125) * 1.250 + 40.0000/2 * 1.125$
 $= 31.094 \text{ in}^2$

Compressive Stress [Sc]:

$= P/Ar$
 $= 13605.3/31.0938 = 437.5584 \text{ psi}$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay ²	Io
Rib	4.0000	8.5938	34.3750	0.0000	53.3333
Web	4.0000	22.5000	90.0000	0.0000	4.7461
Values	4.0000	31.0938	124.3750	0.0000	58.0794

Bending Moment [Rm]:

$$\begin{aligned} &= F1 / (2 * Bplen) * e * r1 / 2 \\ &= 117.2 / (2 * 41.00) * 40.000 * 49.00 / 2 \\ &= 116.738 \text{ ft.lb.} \end{aligned}$$

KL/R < Cc (35.8500 < 126.0993) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = (1 - (35.85)^2 / (2 * 126.10^2)) * 36000 / (5/3 + 3 * (35.85) / (8 * 126.10) - (35.85^3) / (8 * 126.10^3))$$

$$Sca = 19512.54 \text{ psi}$$

AISC Unity Check on Outside Ribs (must be <= 1.0)

$$\text{Check} = Sc/Sca + (Rm/Z)/Sba$$

$$\text{Check} = 437.56/19512.54 + (1400.86/14.520)/24000.00$$

$$\text{Check} = 0.03$$

PV Elite is a trademark of Intergraph CADWorx & Analysis Solutions, Inc. 2015

Input Values:

Wind Design Code		IBC-2009	
Wind Load Reduction Scale Factor		1.0	
Basic Wind Speed	[V]	90.000	mile/hr
Surface Roughness Category		C: Open Terrain	
Importance Factor		1.15	
Type of Surface		Moderately Smooth	
Base Elevation		2.0000	ft.
Percent Wind for Hydrotest		33.0	
Using User defined Wind Press. Vs Elev.		N	
Height of Hill or Escarpment	H or Hh	0.0000	ft.
Distance Upwind of Crest	Lh	0.0000	ft.
Distance from Crest to the Vessel	x	0.0000	ft.
Type of Terrain (Hill, Escarpment)		Flat	
Damping Factor (Beta) for Wind (Ope)		0.0100	
Damping Factor (Beta) for Wind (Empty)		0.0000	
Damping Factor (Beta) for Wind (Filled)		0.0000	

Wind Analysis Results

Static Gust-Effect Factor, Operating Case [G]:

$$\begin{aligned} &= \min(0.85, 0.925((1 + 1.7 * gQ * Izbar * Q) / (1 + 1.7 * gV * Izbar))) \\ &= \min(0.85, 0.925((1+1.7*3.400*0.228*0.980) / (1+1.7*3.400*0.228))) \\ &= \min(0.85, 0.915) \\ &= 0.850 \end{aligned}$$

Natural Frequency of Vessel (Operating)	33.000 Hz
Natural Frequency of Vessel (Empty)	33.000 Hz
Natural Frequency of Vessel (Test)	33.000 Hz

Note: Per Section 1609 of IBC 2003/06/09 these results are also applicable for the determination of Wind Loads on structures (1609.1.1).

User Entered Importance Factor is	1.150
Force Coefficient	[Cf] 0.601
Structure Height to Diameter ratio	7.154

This is classified as a rigid structure. Static analysis performed.

Sample Calculation for the First Element

The ASCE code performs all calculations in Imperial Units only. The wind pressure is therefore computed in these units.

Value of [Alpha] and [Zg]:

Exposure Category: C from Table C6-2
Alpha = 9.500 : Zg = 900.000 ft.

Effective Height [z]:

= Centroid Height + Vessel Base Elevation
= 5.000 + 2.000 = 7.000 ft.

Velocity Pressure coefficient evaluated at height z [Kz]:

Because z (7.000 ft.) < 15 ft.
= 2.01 * (15 / Zg) ^{2 / Alpha}
= 2.01 * (15/900.000) ^{2/9.500}
= 0.849

Type of Hill: No Hill

Wind Directionality Factor [Kd]:
 = 0.95 per [6-6 ASCE-7 98][6-4 ASCE-7 02/05]

As there is No Hill Present: [Kzt]:
 K1 = 0, K2 = 0, K3 = 0

Topographical Factor [Kzt]:
 = (1 + K1 * K2 * K3)²
 = (1 + 0.000 * 0.000 * 0.000)²
 = 1.0000

Velocity Pressure evaluated at height z, Imperial Units [qz]:
 = 0.00256 * Kz * Kzt * Kd * I * Vr (mph)²
 = 0.00256 * 0.849 * 1.000 * 0.950 * 1.150 * 90.000²
 = 19.2 psf

Force on the first element [F]:
 = qz * G * Cf * WindArea
 = 19.231 * 0.850 * 0.601 * 25.977
 = 255.1 lb.

Element	Hgt (z) ft.	K1	K2	K3	Kz	Kzt	qz psf
1203-04 Channel	7.0	0.000	0.000	0.000	0.849	1.000	19.231
1203-04 Shell	7.0	0.000	0.000	0.000	0.849	1.000	19.231

Wind Load Calculation

From	To	Wind Height ft.	Wind Diameter ft.	Wind Area in ²	Wind Pressure psf	Element Wind Load lb.
20	50	7.00000	6.56250	3740.62	19.2307	255.133
50	60	7.00000	6.64062	20400.0	19.2307	1391.40

PV Elite is a trademark of Intergraph CADWorx & Analysis Solutions, Inc. 2015

Input Values:

Earthquake Analysis per IBC 2009

Short-period site coefficient 1613.5.3(1)	Fa: 1.600
Long -period site coefficient 1613.5.3(2)	Fv: 2.400
Maximum Mapped Acceleration Value for Short Periods	Ss: 0.167
Maximum Mapped Acceleration Value for 1 Sec. Period	S1: 0.067
Response Modification Factor	R: 3.000
Importance Factor	Ie: 1.250
Site Class	D

Seismic Analysis Results:

$Sms = Fa * Ss = 1.600 * 0.167 = 0.267$
 $Sm1 = Fv * S1 = 2.400 * 0.067 = 0.161$
 $Sds = 2/3 * Sms = 2/3 * 0.267 = 0.178$

$Sds = \text{Max}(0.8 * Sds, SdsUser)$
 $= \text{Max}(0.143, 0.178)$
 $= 0.178$

$Sd1 = 2/3 * Sm1 = 2/3 * 0.161 = 0.107$

$Sd1 = \text{Max}(0.8 * Sd1, Sd1User)$
 $= \text{Max}(0.086, 0.108)$
 $= 0.108$

Check Approximate Fundamental Period from 12.8-7 [Ta]:

$= Ct * hn^x$ where $Ct = 0.020$, $x = 0.75$ and $hn = \text{Structural Height (ft.)}$
 $= 0.020 * (6.6667^{0.75})$
 $= 0.083 \text{ seconds}$

The Coefficient Cu from Table 12.8-1 is : 1.700

Fundamental Period (1/Frequency) [T]:

$= (1/\text{Natural Frequency}) = (1/33.000)$
 $= 0.030$

Check the Value of T which is the smaller of $Cu * Ta$ and T:

$= \text{Minimum Value of } (1.700 * 0.083, 0.030) \text{ per 12.8.2}$
 $= 0.030$

As the time period is < 0.06 second, use section 15.4.2.

Compute the Base Shear per equation 15.4-5, [V]:

$= 0.3 * Sds * W * I$
 $= 0.3 * 0.178 * 44566 * 1.25$
 $= 2974.789 \text{ lb.}$

Vertical load per 9.5.2.7-1, [YEq]:

$= 0.2 * Sds * W$
 $= 0.2 * 0.178 * 44566 = 1586.55 \text{ lb.}$

Final Base Shear, $V = 2082.35 \text{ lb.}$
Final Vertical Load, $YEq = 1110.59 \text{ lb.}$

Earthquake Load Calculation

From	To	Earthquake Height ft.	Earthquake Weight lb.	Element Ope Load lb.
20	50	1.66667	11141.5	520.588
50	Sad1	1.66667	11141.5	520.588
Sad1	60	1.66667	11141.5	520.588
50	60	1.66667	11141.5	520.588

Note: The Earthquake Loads calculated and printed in the Earthquake Load calculation report have been factored by the input scalar/load reduction factor of: 0.700.

PV Elite is a trademark of Intergraph CADWorx & Analysis Solutions, Inc. 2015